

# Rocket motion fuel consumption

(1.1)

In absence of external forces

$$v(t) = v_0 - \frac{u}{m} \ln\left(\frac{m_0}{m}\right)$$

After all the fuel is used up

$$v_f = v_0 - \frac{u}{m} \ln\left(\frac{m_f}{m_0}\right)$$

$$\rightarrow \frac{v_f - v_0}{u} = - \ln\left(\frac{m_f}{m_0}\right)$$

thus,  $\frac{m_f}{m_0} = e^{-\frac{(v_f - v_0)}{u}}$

$$\text{Fuel used : } M = m_0 - m_f$$

$$= m_0 - m_0 e^{-\frac{(v_f - v_0)}{u}}$$

$$= m_0 \left[ 1 - e^{-\frac{(v_f - v_0)}{u}} \right]$$

thus,  $M = m_0 \left[ 1 - e^{-\frac{(v_f - v_0)}{u}} \right]$



# Rocket motion in a gravitational field

①

$$dm = -dm > 0$$

$u > 0$ : exhaust speed w.r.t spaceship

$$F = \frac{dp}{dt}$$

$$F = -mg \quad (\text{Applied Force})$$

$$t \quad t + dt$$

$$dp = p(t+dt) - p(t)$$

$$= (m-dm')(v+dv) + dm'(v-u) - mv$$

$$= m v - \cancel{v dm'} + mdv - \cancel{dm' dv} + \cancel{dm' v} - u dm' - \cancel{mv}$$

$$= m dv - u dm$$

$$= m dv + u dm$$

$$\text{Thus, } \frac{dp}{dt} = m \frac{dv}{dt} + u \frac{dm}{dt}$$

$$\text{so } \boxed{-mg = m \frac{dv}{dt} + u \frac{dm}{dt}}$$

$$\underline{\text{NOTE:}} \quad m \frac{dv}{dt} = -mg - u \frac{dm}{dt}$$

$$= -mg + \boxed{u \frac{dm'}{dt}}$$

reaction force  
from ejected  
mass  $dm'$

$$\underline{\text{Assume:}} \quad \alpha \equiv -\frac{dm}{dt} (> 0) \text{ is constant}$$

L burn rate

$$\text{Then } dt = -\frac{dm}{\alpha} \rightarrow -mg dt = m dv + u dm$$

$$\frac{mg}{\alpha} dm = m dv + u dm$$

$$dv = \left( \frac{g}{\alpha} - \frac{u}{m} \right) dm$$

Integrale

$$\int_{v_0}^v dv = \int_{m_0}^m \left( \frac{g}{\alpha} - \frac{u}{m} \right) dm$$

$$\boxed{v - v_0 = \frac{g}{\alpha} (m - m_0) - u \ln \left( \frac{m}{m_0} \right)}$$

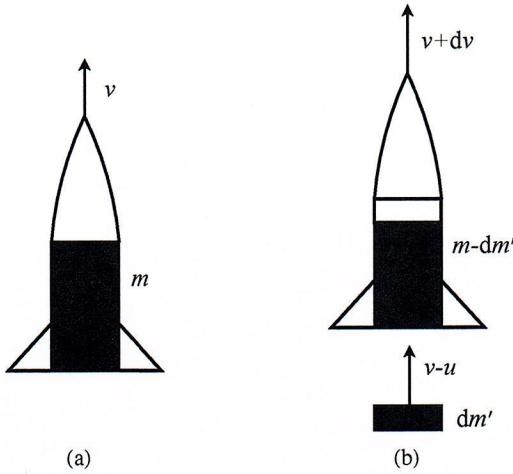
Nun:  $\alpha = -\frac{dm}{dt} \rightarrow dm = -\alpha dt$

$$\int_{m_0}^m dm = -\alpha \int_0^t dt$$

$$m - m_0 = -\alpha t$$

$$\text{so } \boxed{m(t) = m_0 - \alpha t}$$

$$\rightarrow \boxed{v - v_0 = -gt - u \ln \left( 1 - \frac{\alpha t}{m_0} \right)}$$



**Fig. 1.1** A rocket moving in interstellar space, free of all external forces. Panel (a): Rocket at time  $t$  (mass  $m$ , velocity  $v$ ). Panel (b): Rocket and exhaust at time  $t + dt$  (mass  $m - dm'$ , velocity  $v + dv$ ; mass  $dm'$ , velocity  $v - u$ ).

will have changed its velocity to  $v + dv$ . We will assume that the exhaust gases  $dm'$  exit the rocket with *constant* velocity,  $-u$ , with respect to the rocket, so that with respect to the fixed inertial frame, the exhaust gases are moving with velocity  $v - u$ . The change in the momentum of the system over the time interval  $t$  to  $t + dt$  is then

$$\begin{aligned} dp &= p(t + dt) - p(t) \\ &= [(m - dm')(v + dv) + dm'(v - u)] - mv \\ &= mdv - udm' \\ &= mdv + udm, \end{aligned} \tag{1.3}$$

where we ignored the  $-dm'dv$  term (since it is second-order small) to get the third line, and switched back to  $dm$  to get the last line. Since there are no external forces acting on the system,  $dp/dt = F = 0$ , which implies

$$0 = mdv + udm, \tag{1.4}$$

or, equivalently,

$$dv = -u \frac{dm}{m}. \tag{1.5}$$

This is a separable differential equation, which can be immediately integrated, subject to the boundary (or initial) condition that  $v = v_0$  when  $m = m_0$ :

$$v - v_0 = -u \ln(m/m_0). \tag{1.6}$$

### Galilean transformation: (1.3)

$$\vec{r}'(t) = \vec{r}(t) - u t$$

Follows that

$$m \ddot{\vec{r}'} = m \left( \ddot{\vec{r}} - \underbrace{\frac{d^2(u)}{dt^2}(ut)}_{=0} \right)$$

$$= m \ddot{\vec{r}}$$

$$= \vec{F}$$

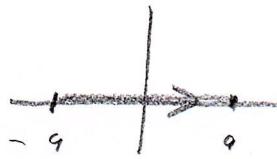
so Newton's 2<sup>nd</sup> law is invariant under such a transformation.

# Work done by velocity-dependent force

(1.4)

$$W_{1,2} = \int_{r_1}^{r_2} \vec{F} \cdot d\vec{s}, \quad \vec{F} = -b \vec{v}$$

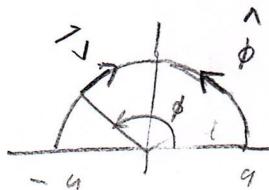
a) For motion



with const speed

$$\begin{aligned} W_{1,2} &= \int_{-a}^a -b v \, dx, \quad v = \text{const} \\ &= -b v \int_{-a}^a dx \\ &= -2abv \end{aligned}$$

b) For motion



with const speed

$$\begin{aligned} W_{1,2} &= \int_{r_1}^{r_2} \vec{F} \cdot d\vec{s}, \quad \vec{F} = -b \vec{v} \hat{\phi}, \quad \vec{v} = v \hat{\phi} \quad \text{where } v > 0 \\ &= \int_0^\pi +b v \hat{\phi} \cdot -a \hat{\phi} \, d\phi \\ &\quad \phi = \pi \\ &= +abv \int_{\pi}^0 d\phi \\ &= -\pi abv \end{aligned}$$

NOTE: Force does more work along second path  
as distance is larger

# Work done by quadratic drag force (1.5)

Find work done in moving the particle a distance  $a$  along a straight line.

$$\rightarrow 1\text{-D motion: } F = -bv^2, \quad dr \rightarrow dx \quad (F = -bv^2 \vec{v})$$

$$W_{12} = \int_0^a F dx = - \int_0^a bv^2 dx$$

$$\text{From: } -bv^2 = F = ma = m \frac{dv}{dt}$$

$$-b v \frac{dx}{dt} = m \frac{dv}{dt}$$

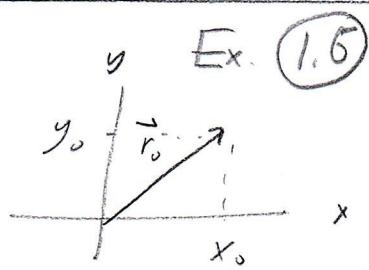
$$\int -\frac{b}{m} dx = \int \frac{dv}{v}$$

$$-\frac{b}{m} x = \ln v$$

$$v(x) = v_0 e^{-\frac{b}{m} x}$$

$$\begin{aligned} \text{Thus, } W_{12} &= - \int_0^a bv^2 dx \\ &= - \int_0^a b v_0^2 e^{-\frac{2b}{m} x} dx \\ &= -b v_0^2 \int_0^a e^{-\frac{2b}{m} x} dx \\ &= -b v_0^2 \left( \frac{m}{-2b} \right) \left( e^{-\frac{2bx}{m}} - 1 \right) \\ &= \frac{1}{2} m v_0^2 \left( e^{-2ba/m} - 1 \right) \end{aligned}$$

Conservation of angular momentum in different coordinate systems (1)



Ex. (1.6)

$$\vec{F} = F \hat{x} \quad (\text{assume } F = \text{const.})$$

$$\vec{F} = m \vec{a} = m \ddot{\vec{r}}$$

$$(a) \quad \ddot{x} = \frac{F}{m} \rightarrow x = x_0 + \vec{x}_0 t + \frac{1}{2} \left( \frac{F}{m} \right) t^2$$

$$\ddot{y} = 0 \rightarrow y = y_0 + \vec{y}_0 t$$

$$\text{Thus, } x(t) = x_0 + \frac{1}{2} \left( \frac{F}{m} \right) t^2$$

$$y(t) = y_0$$

$$\rightarrow \vec{r}(t) = [x_0 + \frac{1}{2} \left( \frac{F}{m} \right) t^2] \hat{x} + y_0 \hat{y}$$

$$= \vec{r}_0 + \frac{1}{2} \left( \frac{F}{m} \right) t^2 \hat{x}$$

$$\text{Also, } \vec{p}(t) = m \dot{\vec{r}}(t)$$

$$= F \hat{x}$$

$$(b) \quad \vec{l} = \vec{r} \times \vec{p}$$

$$= \left[ (x_0 + \frac{1}{2} \left( \frac{F}{m} \right) t^2) \hat{x} + y_0 \hat{y} \right] \times F \hat{x}$$

$$= -F y_0 \hat{z}$$

$$\frac{d\vec{l}}{dt} = -F y_0 \hat{z}$$

$$\vec{\tau} = \vec{r} \times \vec{F}$$

$$= \left[ (x_0 + \frac{1}{2} \left( \frac{F}{m} \right) t^2) \hat{x} + y_0 \hat{y} \right] \times \vec{F}$$

$$= -F y_0 \hat{z} = \frac{d\vec{l}}{dt}$$

Since  $\vec{\tau} \neq 0$ ,  $\frac{d\vec{l}}{dt} \neq 0 \rightarrow \text{Ang. momentum not conserved}$

(2)

(c) New coord system so that

$$\vec{F}' = x_0 \hat{x}$$

Solve Eoms for  $\vec{F} = F \hat{x}$

$$\rightarrow \vec{r}'(t) = \left( x_0 + \frac{1}{2} \left( \frac{F}{m} \right) t^2 \right) \hat{x}$$

$$\vec{p}'(t) = Ft \hat{x}$$

$$\vec{\tau}' = \vec{r}' \times \vec{p}' = 0$$

$$\frac{d\vec{\tau}'}{dt} = 0$$

$$\text{But } \vec{\tau}' = \vec{r}' \times \vec{F}$$

$$= \left( x_0 + \frac{1}{2} \left( \frac{F}{m} \right) t^2 \right) \hat{x} \times F \hat{x}$$

$$= 0$$

$$\text{so } \vec{\tau}' = \frac{d\vec{\tau}'}{dt} = 0$$

Angular momentum conserved in this system,  
but not in the other.

# Expression for total linear momentum

(1.7)

Show:  $\vec{P} = M \vec{R}$

$$M \vec{R} = m \frac{d}{dt} \left( + \sum_{I=1}^n m_I \vec{r}_I \right)$$

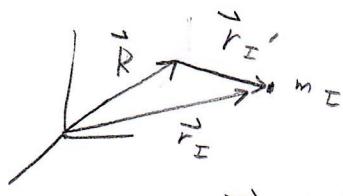
$$= \sum_I m_I \vec{r}_I$$

$$= \sum_I m_I \vec{v}_I$$

$$= \sum_I P_I$$

$$\therefore \vec{P}$$

# Expression for total angular momentum:



$$\vec{r}_I = \vec{R} + \vec{r}'_I$$

Show:  $\vec{L} = \vec{R} \times \vec{p} + \sum_I \vec{r}'_I \times \vec{p}'_I$

Proof: 
$$\begin{aligned} \vec{L} &= \sum_I \vec{r}_I \times \vec{p}_I \\ &= \sum_I \vec{r}_I \times \vec{p}'_I \\ &= \sum_I \vec{r}_I \times m_I \frac{d\vec{r}_I}{dt} \\ &= \sum_I m_I (\vec{R} + \vec{r}'_I) \times \left[ \frac{d\vec{R}}{dt} + \frac{d\vec{r}'_I}{dt} \right] \\ &= \sum_I m_I \left[ \vec{R} \times \frac{d\vec{R}}{dt} + \vec{r}'_I \times \frac{d\vec{r}'_I}{dt} \right. \\ &\quad \left. + \vec{R} \times \frac{d\vec{r}'_I}{dt} + \vec{r}'_I \times \frac{d\vec{R}}{dt} \right] \\ &= M \vec{R} \times \frac{d\vec{R}}{dt} + \sum_I \vec{r}'_I \times \left( m_I \frac{d\vec{r}'_I}{dt} \right) \\ &\quad + M \vec{R} \times \sum_I m_I \frac{d\vec{r}'_I}{dt} + \left( \sum_I m_I \vec{r}'_I \right) \times \frac{d\vec{R}}{dt} \\ &= \vec{R} \times \vec{p} + \sum_I \vec{r}'_I \times \vec{p}'_I + \vec{R} \times \frac{d}{dt} \left( \sum_I m_I \vec{r}'_I \right) \\ &\quad + \left( \sum_I m_I \vec{r}'_I \right) \times \frac{d\vec{R}}{dt} \quad \stackrel{\text{see below}}{=} \\ &= \vec{R} \times \vec{p} + \sum_I \vec{r}'_I \times \vec{p}'_I \end{aligned}$$

NOTE: 
$$\begin{aligned} \vec{R} &= \frac{1}{M} \sum_I m_I \vec{r}_I \\ \vec{r}'_I &= \vec{r}_I - \vec{R} \end{aligned} \quad \Rightarrow \quad \sum_I m_I \vec{r}'_I = \sum_I m_I (\vec{r}_I - \vec{R})$$

$$\begin{aligned} &= \sum_I m_I \vec{r}_I - \left( \sum_I m_I \right) \vec{R} \\ &= M \vec{R} - M \vec{R} = 0 \end{aligned}$$

Exer ⑧

$$\text{Velocity } \frac{d\vec{r}}{dt} = \vec{v}(e)$$

$$\vec{L} = \vec{r}_I \times \vec{M}$$

$$\frac{d\vec{L}}{dt} = \vec{M} \frac{d\vec{r}_I}{dt}$$

$$= \vec{M} \frac{d}{dt} (\vec{r}_I \times \vec{p}_I)$$

$$= \vec{M} \left( \vec{v}_I \times \vec{p}_I + \vec{r}_I \times \dot{\vec{p}}_I \right)$$

since  $\vec{p}_I = m \vec{v}_I$

$$= \vec{M} \vec{r}_I \times \vec{F}_I$$

$$= \vec{M} \vec{r}_I \times \left( \vec{F}_I^{(e)} + \sum_{J \neq I} \vec{F}_{JI} \right)$$

$$= \vec{M} \vec{r}_I \times \vec{F}_I^{(e)} + \sum_{J \neq I} \vec{M} \vec{r}_I \times \vec{F}_{JI}$$

$$= \vec{M} \vec{r}_I^{(e)} + \sum_{I,J} \left( \vec{r}_I \times \vec{F}_{JI} + \vec{r}_J \times \vec{F}_{IJ} \right) - \vec{M} \vec{r}_I$$

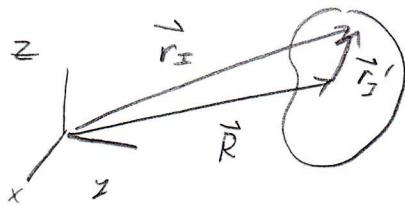
$$= \vec{M} \vec{r}_I^{(e)} + \frac{1}{2} \sum_{I,J} \left( \vec{r}_I - \vec{r}_J \right) \times \vec{F}_{JI}$$

$$\cancel{\theta(\vec{F}_I - \vec{r}_J)}$$

$$= \vec{M} \vec{r}_I^{(e)}$$

$$= \vec{r}_I^{(e)}$$

Euler 1.9



$$\begin{aligned} \vec{L} &= \sum_{\text{H}} \vec{r}_{\text{E}} \times \vec{p}_{\text{E}} \\ &= \sum_{\text{H}} (\vec{R} + \vec{r}_{\text{E}}^{\text{I}'}) \times \vec{p}_{\text{E}} \\ &= \vec{R} \times \vec{p} + \sum_{\text{H}} \vec{r}_{\text{E}}^{\text{I}'} \times \vec{p}_{\text{E}} \end{aligned}$$

Now:  $\vec{p}_{\text{E}} = m_{\text{I}} \vec{v}_{\text{I}} = m_{\text{I}} \dot{\vec{r}}_{\text{I}}$

$$\begin{aligned} &= m_{\text{I}} (\vec{R} + \vec{r}_{\text{E}}^{\text{I}'}) \circ \\ &= m_{\text{I}} \vec{R} \circ + m_{\text{I}} \vec{r}_{\text{E}}^{\text{I}'} \end{aligned}$$

Thus,  $\vec{L} = \vec{R} \times \vec{p} + \sum_{\text{I}} m_{\text{I}} \vec{r}_{\text{E}}^{\text{I}'} \times (m_{\text{I}} \vec{R} + m_{\text{I}} \vec{r}_{\text{E}}^{\text{I}'})$

$$= \vec{R} \times \vec{p} + \left( \sum_{\text{I}} m_{\text{I}} \vec{r}_{\text{E}}^{\text{I}'} \right) \times \vec{R} + \sum_{\text{I}} \vec{r}_{\text{E}}^{\text{I}'} \times \vec{p}_{\text{E}}^{\text{I}'}$$

Since  $\vec{R}$  is com

$$= \vec{R} \times \vec{p} + \sum_{\text{I}} \vec{r}_{\text{E}}^{\text{I}'} \times \vec{p}_{\text{E}}^{\text{I}'}$$

Torque equation for example problem: (1.10)

$$\begin{aligned}\vec{\tau} &= \sum_I \vec{\tau}_I = \vec{r}_1 \times (f \hat{z} \times \vec{r}_{12}) + \vec{r}_2 \times (f \hat{z} \times \vec{r}_{21}) \\ &= 2\vec{r}_1 \times (f \hat{z} \times \vec{r}_{12}) - \vec{r}_1 \cdot -\vec{r}_{12} \\ &= 4f \vec{r} \times (\hat{z} \times \vec{r}) \quad \text{using } \vec{r}_{12} = 2\vec{r}_1 \approx 2\vec{r} \\ &= 4f [\hat{z}(\vec{r} \cdot \vec{r}) - \vec{r}(\vec{r} \cdot \hat{z})] \\ &= 4f r^2 \hat{z}\end{aligned}$$

° since  $\vec{r}$  is in  $(x, y)$  plane

Now:  ~~$\omega$~~   $\omega = \sqrt{\frac{f}{m}} \rightarrow f = m\omega^2$

$$\begin{aligned}\text{Also: } r^2 &= x^2 + y^2 \\ &= ch^2 c^2 + sh^2 s^2 \\ &= ch^2 (1-s^2) + sh^2 s^2 \\ &= ch^2 - s^2 \underbrace{(ch^2 - sh^2)}_{=1} \\ &= ch^2 - s^2 \\ &= \left( \frac{1 + \cosh(2wt)}{2} \right) - \left( \frac{1 - \cos(2wt)}{2} \right) \\ &= \frac{1}{2} (\cosh(2wt) + \cos(2wt))\end{aligned}$$

$$\begin{aligned}\rightarrow \vec{\tau} &= 4m\omega^2 \frac{1}{2} (\cosh(2wt) + \cos(2wt)) \hat{z} \\ &= 2m\omega^2 (\cosh(2wt) + \cos(2wt)) \hat{z}\end{aligned}$$

Recall:

$$\vec{L} = m\omega^2 w [\sinh(2wt) + \sinh(2wt)] \hat{z}$$

$$\begin{aligned}\rightarrow \frac{d\vec{L}}{dt} &= 2m\omega^2 w^2 [\cos(2wt) + \cosh(2wt)] \hat{z} \\ &= \vec{\tau}\end{aligned}$$

Time rate of change of work for example problem: (1.11) ①

$$\begin{aligned}\frac{dw}{dt} &= 4 F (\vec{z} \times \vec{v}_i) \cdot \vec{v}_i \\ &= 4 F (\vec{r}_i \times \vec{v}_i) \cdot \vec{z}' \\ &= 4 m \omega^2 (\vec{r}_i \times \vec{v}_i) \cdot \vec{z}' \\ &= 4 \omega^2 (\vec{r}_i \times \vec{p}_i) \cdot \vec{z}' \\ &= 4 \omega^2 \vec{L} \cdot \vec{z}' \\ &= 2 \omega^2 \vec{L} \cdot \vec{z}' \quad (\text{since } \vec{L} \text{ points along } \vec{z}) \\ &= 2 \omega^2 L\end{aligned}$$

Integrate this:

$$\begin{aligned}dw &= 2 \omega^2 L dt \\ &= 2 \omega^2 (m a^2 \omega) [\sin(2\omega t) + \sinh(2\omega t)] dt\end{aligned}$$

$$\begin{aligned}\rightarrow w &= \cancel{\frac{1}{2}} m a^2 \omega^3 \left[ \frac{1}{2\omega} \sin(2\omega t) + \frac{\cosh(2\omega t)}{2\omega} \right] \\ &= m a^2 \omega^2 [\cosh(2\omega t) - \cos(2\omega t)] \\ &= F a^2 [\cosh(2\omega t) - \cos(2\omega t)]\end{aligned}$$

$$\begin{aligned}T &= \sum_{i=1}^{\infty} \frac{1}{2} m_i \vec{v}_i^2 \\ &= \cancel{\frac{1}{2}} \cdot \left( \cancel{\frac{1}{2}} m \vec{v}_i^2 \right) \\ &= m (\dot{x}^2 + \dot{y}^2) \\ &= m a^2 \omega^2 [(-s_{gh} + c_{sh})^2 + (c_{sh} + s_{ch})^2] \\ &= m a^2 \omega^2 [s^2 ch^2 + c^2 sh^2 - \cancel{2s c \cancel{ch} sh} + c^2 sh^2 + s^2 ch^2 + \cancel{2s c \cancel{ch} sh}] \\ &= 2 m a^2 \omega^2 [s^2 ch^2 + c^2 sh^2] \\ &= 2 m a^2 \omega^2 [(1 - c^2) ch^2 + c^2 sh^2] \\ &= 2 m a^2 \omega^2 [ch^2 - \underbrace{c^2 (ch^2 - sh^2)}_1]\end{aligned}$$

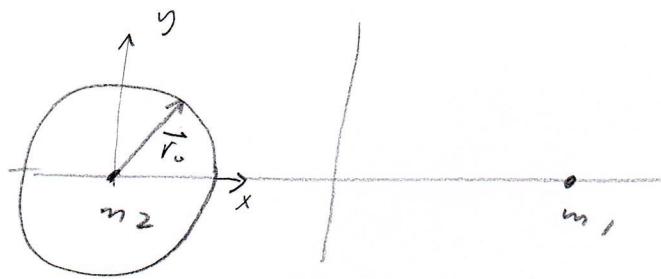
(2)

$$\begin{aligned}&= 2 m a^2 \omega^2 [c b^2 - c^2] \\&= 2 m a^2 \omega^2 \left[ \left( \frac{1 + \cosh(2\omega t)}{2} \right) - \left( \frac{1 + \cos(2\omega t)}{2} \right) \right] \\&= m a^2 \omega^2 [\cosh(2\omega t) - \cos(2\omega t)] \\&= f a^2 [\cosh(2\omega t) - \cos(2\omega t)] \\&= w\end{aligned}$$

# No h-conservative force for example problem

(1,12)

$\Gamma$ ,  $R_0$  origin centered on  $m_2$



$$\vec{F} = f \hat{z} \times \vec{r}_0 \\ = f r_0 \hat{\phi}$$

$$\oint_C \vec{F} \cdot d\vec{r} = \int_{\phi=0}^{2\pi} \vec{F} \cdot r_0 d\phi \hat{\phi} \\ = \int_{\phi=0}^{2\pi} (f r_0 \hat{\phi}) \cdot r_0 d\phi \hat{\phi} \\ = f r_0^2 \int_{\phi=0}^{2\pi} d\phi \\ = 2\pi f r_0^2 \\ \neq 0$$

so  $\vec{F}$  is not conservative

# Instantaneous angular velocity vector:

1.13

Let

$$\vec{A}_{ij} = \begin{vmatrix} \cos \omega t & \sin \omega t & 0 \\ -\sin \omega t & \cos \omega t & 0 \\ 0 & 0 & 1 \end{vmatrix}$$

$$\vec{A}'_{ij} = \omega \begin{vmatrix} -\sin & \cos & 0 \\ -\cos & -\sin & 0 \\ 0 & 0 & 0 \end{vmatrix}$$

$$\text{Now } M_{j|i} = \sum_{i'} \vec{A}_{ij} \vec{A}'_{i'i} = \vec{A}^T \vec{A}'$$

$$= \omega \begin{vmatrix} C & -S & 0 \\ +S & C & 0 \\ 0 & 0 & 1 \end{vmatrix} \begin{vmatrix} -S & C & 0 \\ -C & -S & 0 \\ 0 & 0 & 0 \end{vmatrix}$$

$$= \omega \begin{vmatrix} 0 & 1 & 0 \\ -1 & 0 & 0 \\ 0 & 0 & 0 \end{vmatrix}$$

$$\boxed{M_{12} = \omega, M_{21} = -\omega, \text{ all other } = 0}$$

$$w_i = \frac{1}{2} \sum_{j,i} \epsilon_{ijk} M_{j|i}$$

$$\rightarrow w_1 = \frac{1}{2} \sum_{j,i} \epsilon_{1ji} M_{j|i} = \frac{1}{2} (\epsilon_{123} \vec{M}_{23}^{\circ} + \epsilon_{132} \vec{M}_{32}^{\circ}) = 0$$

$$w_2 = \frac{1}{2} \sum_{j,i} \epsilon_{2ji} M_{j|i} = \frac{1}{2} (\epsilon_{231} \vec{M}_{31}^{\circ} + \epsilon_{213} \vec{M}_{13}^{\circ}) = 0$$

$$w_3 = \frac{1}{2} \sum_{j,i} \epsilon_{3ji} M_{j|i} = \frac{1}{2} (\epsilon_{312} M_{12} + \epsilon_{321} M_{21}) \\ = \frac{1}{2} (1 \cdot \omega + (-1)(-\omega)) \\ = \omega$$

$$\boxed{\vec{w} = \omega \hat{z}}$$

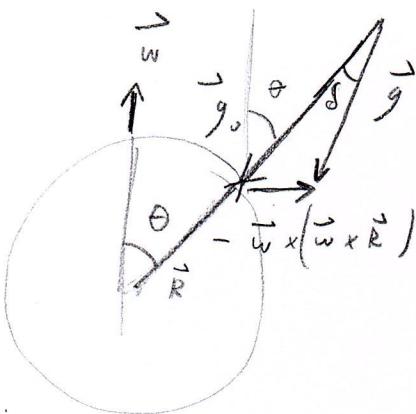
# (1)

Deflection of plumb line away from  $\vec{g}_0$  : (1.14)

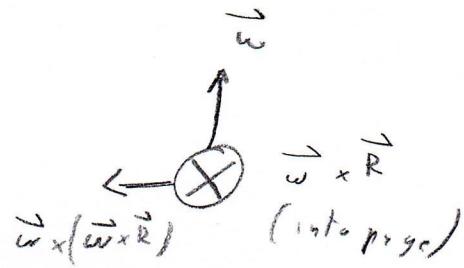
$$R = 6400 \text{ km} = 6.4 \times 10^7 \text{ m}$$

$$\omega = \frac{2\pi}{1 \text{ day}} = \frac{2\pi}{86400 \text{ s}} = 7.27 \times 10^{-5} \frac{\text{rad}}{\text{s}}$$

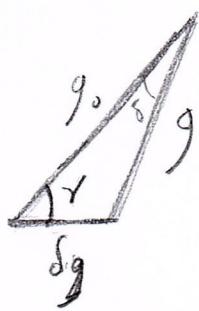
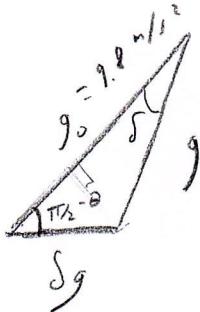
$$g = 9.8 \text{ m/s}^2$$



$$|\vec{\omega} \times (\vec{\omega} \times \vec{R})| = \omega \sin \theta \cdot R \\ = \omega^2 \sin \theta \cdot R$$



$$= 0.3385 \sin \theta \frac{\text{m}}{\text{s}^2} \quad = |\delta \vec{g}| \quad (\text{about } 3\% \text{ of } g_0)$$



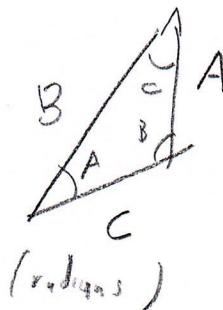
$$(1) \alpha = (2)(\pi/2 - \theta) \\ = (2)(\pi/2) \cos \theta + \sin(\pi/2) \sin \theta$$

$$g^2 = (\delta g)^2 + g_0^2 \\ - 2 \delta g g_0 \cos \alpha$$

$$\delta g = (\delta g)^2 + g_0^2 + 2 \delta g g_0 \sin \theta$$

In w of 1.14

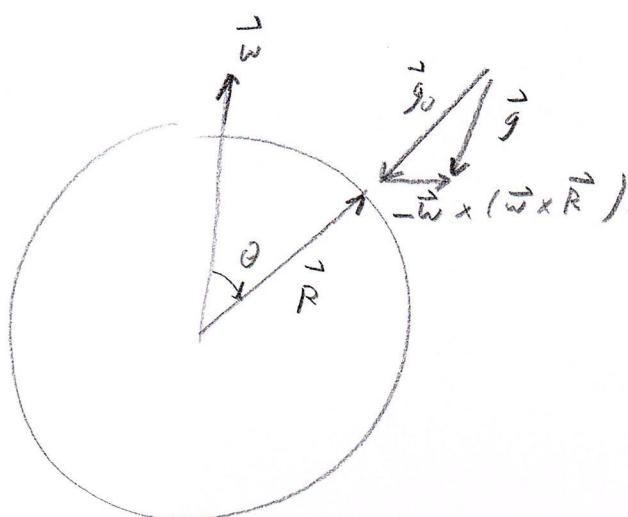
$$\frac{\sin A}{A} = \frac{\sin B}{B} = \frac{\sin C}{C}$$



$$\sim \text{Therefore } \delta \approx \frac{\delta g}{g_0} = \frac{0.3385 \sin \theta}{9.8} = 0.0345 \sin \theta \quad (\text{in radians})$$

$$\frac{1}{9.8} \approx \frac{1}{10} = 0.03$$

Deflection of plumb line away from  $\vec{g}_0$  (1,14)

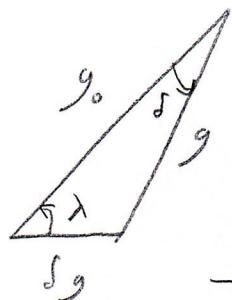
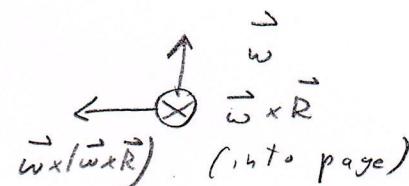


$$R = 6400 \text{ km} \\ = 6.4 \times 10^6 \text{ m}$$

$$\omega = \frac{2\pi}{day} = \frac{2\pi}{86400s} \\ = 7.27 \times 10^{-5} \frac{\text{rad}}{\text{sec}}$$

$$g_0 = 9.8 \text{ m/s}^2$$

$$|-\vec{\omega} \times (\vec{\omega} \times \vec{R})| = \omega \sin \theta \omega R \\ = \omega^2 \sin \theta R \\ = 0.03385 \sin \theta \frac{\text{m}}{\text{s}^2} \\ = |\delta \vec{g}| \quad (\text{about } 0.3\% \text{ of } g_0)$$



$$\lambda = \frac{\pi}{2} - \theta \quad (\text{known})$$

$$\delta g = \omega^2 R \cos \theta \quad (\text{known})$$

$$g_0 = 9.8 \text{ m/s}^2 \quad (\text{known})$$

$$\rightarrow g^2 = g_0^2 + \delta g^2 - 2 g_0 \delta g \cos \lambda \quad (\text{law of cosines}) \\ = g_0^2 + \delta g^2 - 2 g_0 \delta g \sin \theta$$

$$\cos \lambda = \cos \left( \frac{\pi}{2} - \theta \right)$$

$$= \cos \left( \frac{\pi}{2} \right) \cos \theta + \sin \left( \frac{\pi}{2} \right) \sin \theta$$

$$= \sin \theta$$

$$\text{Thus, } g = \sqrt{g_0^2 + \delta g^2 - 2 g_0 \delta g \sin \theta} \\ = g_0 \sqrt{1 - 2 \left( \frac{\delta g}{g_0} \right) \sin \theta + \left( \frac{\delta g}{g_0} \right)^2} \\ \approx g_0 \left[ 1 - \left( \frac{\delta g}{g_0} \right) \sin \theta \right]$$

(2)

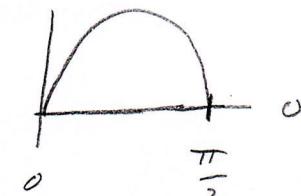
$$\frac{\sin \delta}{\delta_g} = \frac{\sin \lambda}{g} \quad (\text{law of sines})$$

small angle approx:  $\sin \delta \approx \delta$

Thus

$$\begin{aligned} \delta &\approx \delta_g \cdot \frac{\sin \lambda}{g} \\ &\approx \frac{\delta_g}{g_0} \cos \theta \quad (\text{to leading order}) \\ &= \frac{\omega^2 R \sin \theta \cos \theta}{g_0} \\ &= \frac{1}{2} \frac{\omega^2 R \sin 2\theta}{g_0} \end{aligned}$$

$$\begin{aligned} \sin \lambda &= \sin \left( \frac{\pi}{2} - \theta \right) \\ &= \sin \frac{\pi}{2} \cos \theta - \cos \frac{\pi}{2} \sin \theta \\ &= \cos \theta \end{aligned}$$



$$\delta_{\max} \text{ at } \theta = \frac{\pi}{4}$$

$$\begin{aligned} \delta \Big|_{\max} &= \frac{1}{2} \frac{\omega^2 R \sin \left( \frac{\pi}{4} \right)}{g_0} \\ &= \frac{1}{2} \frac{\omega^2 R}{g_0} = \boxed{0.0017} \quad \left( \frac{180^\circ}{\pi} \right) = 0.099^\circ \\ &\approx \boxed{0.1^\circ} \end{aligned}$$

$$\begin{aligned} \delta_g &= \omega^2 R \sin \theta \\ &= 0.03385 \sin \theta \text{ m/s}^2 \end{aligned}$$

Max at equator ( $\pi/2$ )

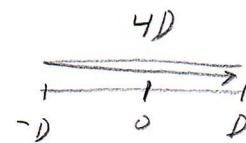
$$\left( \frac{\delta_g}{g_0} \right)_{\max} = \frac{0.03385}{9.8} = \boxed{0.00345} \approx 0.3\%$$

Verify that  $|\omega_x j| \ll g$  (1.15)

$$\omega_x \approx -\omega \sin \theta$$

$$|\omega_x| \leq \omega$$

$$j \sim \frac{4D}{P}$$



$$\text{where } P = \frac{2\pi}{\omega} = \frac{2\pi}{\sqrt{g/L}} \Rightarrow \frac{L}{g} = \frac{P^2}{4\pi^2}$$

$$\text{Thus, } \frac{|\omega_x j|}{g} \leq \frac{\omega(4D)}{P} \frac{1}{g}$$

$$= 4 \left( \frac{\omega}{P} \right) \left( \frac{D}{L} \right) \frac{1}{g}$$

$$= 4 \left( \frac{D}{L} \right) \left( \frac{2\pi}{1 \text{ day}} \right) \frac{P}{4\pi^2} \frac{1}{g}$$

$$= 4 \left( \frac{D}{L} \right) \left( \frac{1}{2\pi} \right) \left( \frac{P}{1 \text{ day}} \right)$$

$$= \frac{2}{\pi} \left( \frac{D}{L} \right) \left( \frac{P}{1 \text{ day}} \right)$$

$$\underset{\ll 1}{\text{w}}, \underset{\ll 1}{\text{w}}, \underset{\ll 1}{\text{P}}$$



$$\left| \begin{aligned} P &= \frac{2\pi}{\sqrt{g/L}} = \frac{2\pi}{\sqrt{10/30}} \\ &\approx 11 \text{ s} \end{aligned} \right.$$

(1)

Foucault pendulum solution ; 1.16

$$\ddot{s}(t) = A e^{\lambda_+ t} + B e^{\lambda_- t}$$

$$\text{where } \lambda_{\pm} = -i(\omega_z \mp \Omega)$$

$$\rightarrow \boxed{\ddot{s}(t) = A e^{-i\omega_z t} e^{int} + B e^{-i\omega_z t} e^{-int}} \\ = [A e^{int} + B e^{-int}] e^{-i\omega_z t}$$

Initial conditions:

$$s(0) = D, \dot{s}(0) = 0$$

$$\boxed{D = s(0) = A + B} \quad (1)$$

$$\ddot{s}(t) = \lambda_+ A e^{\lambda_+ t} + \lambda_- B e^{\lambda_- t}$$

$$\begin{aligned} \rightarrow \dot{s}(0) &= \lambda_+ A + \lambda_- B \\ &= -i(\omega_z - \Omega) A - i(\omega_z + \Omega) B \\ &= -i\omega_z (A + B) + i\Omega (A - B) \end{aligned}$$

$$\text{thus, } \boxed{0 = -i\omega_z (A + B) + i\Omega (A - B)} \quad (2)$$

Substituting (1) into (2) gives

$$0 = -i\omega_z D + i\Omega (A - B)$$

$$i\omega_z D = i\Omega (A - B)$$

$$A - B = \left(\frac{\omega_z}{\Omega}\right) D \Rightarrow \boxed{\begin{aligned} A &= \frac{1}{2} \left(1 + \frac{\omega_z}{\Omega}\right) D \\ B &= \frac{1}{2} \left(1 - \frac{\omega_z}{\Omega}\right) D \end{aligned}}$$

$$\text{and } A + B = D$$

Therefore,

(2)

$$\begin{aligned} \zeta(t) &= \left[ \frac{1}{2} \left( 1 + \frac{\omega_2}{\sqrt{n}} \right) D e^{int} + \frac{1}{2} \left( 1 - \frac{\omega_2}{\sqrt{n}} \right) D e^{-int} \right] e^{-i\omega_2 t} \\ &= D \left[ \underbrace{\frac{1}{2} (e^{int} + e^{-int})}_{\text{const}} + \underbrace{\frac{1}{2} \left( \frac{\omega_2}{\sqrt{n}} \right) (e^{int} - e^{-int})}_{\left( \frac{\omega_2}{\sqrt{n}} \right) i \sin nt} \right] e^{-i\omega_2 t} \\ &= D \left[ \text{const} + i \left( \frac{\omega_2}{\sqrt{n}} \right) \sin nt \right] e^{-i\omega_2 t} \\ &= D \left[ \text{const} + i \left( \frac{\omega_2}{\sqrt{n}} \right) \sin nt \right] (\cos(\omega_2 t) - i \sin(\omega_2 t)) \\ &= D \left[ \cos(\omega_2 t) \cos(\omega_2 t) + \left( \frac{\omega_2}{\sqrt{n}} \right) \sin(nt) \cos(\omega_2 t) \right] \\ &\quad + i D \left[ -\cos(\omega_2 t) \sin(\omega_2 t) + \left( \frac{\omega_2}{\sqrt{n}} \right) \sin(nt) \cos(\omega_2 t) \right] \end{aligned}$$

$$\rightarrow \boxed{\begin{aligned} x(t) &= D \left[ \cos(\omega_2 t) \cos(\omega_2 t) + \frac{\omega_2}{\sqrt{n}} \sin(nt) \sin(\omega_2 t) \right] \\ y(t) &= D \left[ -\sin(\omega_2 t) \cos(nt) + \frac{\omega_2}{\sqrt{n}} \cos(\omega_2 t) \sin(nt) \right] \end{aligned}}$$

Time derivatives of basis vectors for spherical pendulum example:

Use  $\frac{d\hat{e}}{dt} = \frac{\partial \hat{e}}{\partial r} \dot{r} + \frac{\partial \hat{e}}{\partial \theta} \dot{\theta} + \frac{\partial \hat{e}}{\partial \phi} \dot{\phi}$

(1.17)

From App A:

$$\frac{\partial \hat{r}}{\partial r} = 0, \quad \frac{\partial \hat{r}}{\partial \theta} = \hat{\theta}, \quad \frac{\partial \hat{r}}{\partial \phi} = \sin \theta \hat{\phi}$$

$$\rightarrow \boxed{\frac{d\hat{r}}{dt} = \dot{\theta} \hat{\theta} + \sin \theta \dot{\phi} \hat{\phi}}$$

From App A:

$$\frac{\partial \hat{\theta}}{\partial r} = 0, \quad \frac{\partial \hat{\theta}}{\partial \theta} = -\hat{r}, \quad \frac{\partial \hat{\theta}}{\partial \phi} = \cos \theta \hat{\phi}$$

$$\rightarrow \boxed{\frac{d\hat{\theta}}{dt} = -\dot{\theta} \hat{r} + \cos \theta \dot{\phi} \hat{\phi}}$$

From App A:

$$\frac{\partial \hat{\phi}}{\partial r} = 0, \quad \frac{\partial \hat{\phi}}{\partial \theta} = 0, \quad \frac{\partial \hat{\phi}}{\partial \phi} = -\sin \theta \hat{r} - \cos \theta \hat{\theta}$$

$$\rightarrow \boxed{\frac{d\hat{\phi}}{dt} = -\sin \theta \dot{\phi} \hat{r} - \cos \theta \dot{\phi} \hat{\theta}}$$

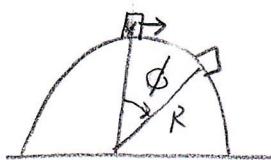
Relativistic problem allowing for initial velocity  $v_0$

①

118

$v_0$

Same analysis as before:



$$\frac{1}{2} \dot{\phi}^2 = -\frac{g}{R} \cos \phi + C$$

C determined from the requirement that

$$v_0 = R\dot{\phi} \text{ at } \phi = 0.$$

$$\rightarrow \frac{1}{2} \left( \frac{v_0}{R} \right)^2 = -\frac{g}{R} \cos 0 + C$$

$$C = \frac{1}{2} \left( \frac{v_0}{R} \right)^2 + \frac{g}{R}$$

$$\text{Thus, } \frac{1}{2} \dot{\phi}^2 = \frac{g}{R} (1 - \cos \phi) + \frac{1}{2} \left( \frac{v_0}{R} \right)^2$$

Equation for normal force:

$$\begin{aligned} F_n &= mg \cos \phi - m R \dot{\phi}^2 \\ &= mg \cos \phi - m R \left[ \frac{2g}{R} (1 - \cos \phi) + \left( \frac{v_0}{R} \right)^2 \right] \\ &= mg \cos \phi - 2mg(1 - \cos \phi) - \frac{m v_0^2}{R} \\ &= mg [3 \cos \phi - 2] - \frac{m v_0^2}{R}. \end{aligned}$$

$$F_n = 0 \Leftrightarrow mg [3 \cos \phi - 2] = \frac{m v_0^2}{R}$$

$$3 \cos \phi - 2 = \frac{v_0^2}{Rg}$$

$$\boxed{\cos \phi = \frac{2}{3} + \frac{1}{3} \frac{v_0^2}{Rg}}$$

(2)

Minimum velocity needed for string to leave surface at  $\phi = 0$ :

$$\cos \theta = \frac{2}{3} + \frac{1}{3} \frac{v_0^2}{R_y}$$

$$1 = \frac{2}{3} + \frac{1}{3} \frac{v_0^2}{R_y}$$

$$\frac{1}{3} = \frac{1}{3} \frac{v_0^2}{R_y}$$

$$\boxed{v_0 = \sqrt{R_y}}$$

# Acceleration Vector in sph. polar coords.

(1)

Prob 1.1

$$\vec{F} = r\hat{r}$$

Velocity:

$$\vec{v} \equiv \frac{d\vec{r}}{dt} = \dot{r}\hat{r} + r\frac{d\hat{r}}{dt}$$

$$\text{Now: } \frac{d\hat{r}}{dt} = \frac{\partial \hat{r}}{\partial r} \dot{r} + \frac{\partial \hat{r}}{\partial \theta} \dot{\theta} + \frac{\partial \hat{r}}{\partial \phi} \dot{\phi}$$

Use results from App A:

$$\frac{\partial \hat{r}}{\partial r} = 0, \quad \frac{\partial \hat{r}}{\partial \theta} = \hat{\theta}, \quad \frac{\partial \hat{r}}{\partial \phi} = \sin\theta \hat{\phi}$$

$$\rightarrow \frac{d\hat{r}}{dt} = \dot{\theta}\hat{\theta} + \sin\theta \dot{\phi}\hat{\phi}$$

$$\text{Thus, } \boxed{\vec{v} = r\hat{r} + r\dot{\theta}\hat{\theta} + r\sin\theta \dot{\phi}\hat{\phi}}$$

Acceleration:

$$\begin{aligned} \vec{a} &\equiv \frac{d\vec{v}}{dt} \\ &= \ddot{r}\hat{r} + \dot{r}\frac{d\hat{r}}{dt} + \dot{r}\dot{\theta}\hat{\theta} + r\ddot{\theta}\hat{\theta} + r\dot{\theta}\frac{d\hat{\theta}}{dt} \\ &\quad + \dot{r}\sin\theta \dot{\phi}\hat{\phi} + r\cos\theta \dot{\theta}\dot{\phi}\hat{\phi} + r\sin\theta \dot{\phi}\dot{\phi}\hat{\phi} \\ &\quad + r\sin\theta \dot{\phi}\frac{d\hat{\phi}}{dt} \end{aligned}$$

$$\frac{d\hat{r}}{dt} = \dot{\theta}\hat{\theta} + \sin\theta \dot{\phi}\hat{\phi} \quad (\text{as before})$$

$$\frac{d\hat{\theta}}{dt} = \frac{\partial \hat{\theta}}{\partial r} \dot{r} + \frac{\partial \hat{\theta}}{\partial \theta} \dot{\theta} + \frac{\partial \hat{\theta}}{\partial \phi} \dot{\phi}$$

From App A:

$$\frac{\partial \hat{\theta}}{\partial r} = 0, \quad \frac{\partial \hat{\theta}}{\partial \theta} = -\hat{r}, \quad \frac{\partial \hat{\theta}}{\partial \phi} = \cos\theta \hat{\phi}$$

(2)

$$\rightarrow \frac{d\hat{\theta}}{dt} = -\dot{\theta}\hat{r} + \omega\theta\hat{\phi}\hat{\phi}$$

Also,

$$\frac{d\hat{\phi}}{dt} = \frac{\partial \hat{\phi}}{\partial r} \dot{r} + \frac{\partial \hat{\phi}}{\partial \theta} \dot{\theta} + \frac{\partial \hat{\phi}}{\partial \phi} \dot{\phi}$$

From App A:

$$\frac{\partial \hat{\phi}}{\partial r} = 0, \quad \frac{\partial \hat{\phi}}{\partial \theta} = 0, \quad \frac{\partial \hat{\phi}}{\partial \phi} = -\sin\theta \hat{r} - \cos\theta \hat{\theta}$$

$$\rightarrow \frac{d\hat{\phi}}{dt} = -\sin\theta \dot{\phi} \hat{r} - \cos\theta \dot{\phi} \hat{\theta}$$

Thus,

$$\begin{aligned} \vec{q} &= \dot{r}\hat{r} + \dot{r}(\dot{\theta}\hat{\theta} + \cos\theta\dot{\phi}\hat{\phi}) + \dot{r}\dot{\theta}\hat{\theta} + r\ddot{\theta}\hat{\theta} \\ &\quad + r\dot{\theta}(-\dot{\theta}\hat{r} + \cos\theta\dot{\phi}\hat{\phi}) + r\sin\theta\dot{\phi}\hat{\phi} \\ &\quad + r\cos\theta\dot{\theta}\hat{\phi} + r\sin\theta\dot{\phi}\hat{\phi} \\ &\quad + r\sin\theta\dot{\phi}(-\sin\theta\dot{\phi}\hat{r} - \cos\theta\dot{\phi}\hat{\theta}) \end{aligned}$$

$$\begin{aligned} &= \hat{r} [ \ddot{r} - r\dot{\theta}^2 - r\sin^2\theta\dot{\phi}^2 ] \\ &\quad + \hat{\theta} [ 2\dot{r}\dot{\theta} + r\ddot{\theta} - r\sin\theta\cos\theta\dot{\phi}^2 ] \\ &\quad + \hat{\phi} [ 2\sin\theta\dot{r}\dot{\phi} + 2r\cos\theta\dot{\theta}\dot{\phi} + r\sin\theta\ddot{\phi} ] \end{aligned}$$

Acceleration vector in cylindrical coords  $(\rho, \phi, z)$  (Prob 1.2) ①

$$\vec{r} = \hat{\rho}\hat{\rho} + z\hat{z}$$

Velocity:

$$\begin{aligned}\vec{v} &= \frac{d\vec{r}}{dt} \\ &= \dot{\rho}\hat{\rho} + \rho \frac{d\hat{\rho}}{dt} + \dot{z}\hat{z} + z \frac{d\hat{z}}{dt} \\ &= \dot{\rho}\hat{\rho} + \dot{z}\hat{z} + \rho \frac{d\hat{\rho}}{dt}\end{aligned}$$

$$\text{Now: } \frac{d\hat{\rho}}{dt} = \frac{\partial \hat{\rho}}{\partial \rho} \dot{\rho} + \frac{\partial \hat{\rho}}{\partial \phi} \dot{\phi} + \frac{\partial \hat{\rho}}{\partial z} \dot{z}$$

From App A:

$$\frac{\partial \hat{\rho}}{\partial \rho} = 0, \quad \frac{\partial \hat{\rho}}{\partial \phi} = \hat{\phi}, \quad \frac{\partial \hat{\rho}}{\partial z} = 0$$

$$\rightarrow \frac{d\hat{\rho}}{dt} = \dot{\phi}\hat{\phi}$$

$$\text{so } \boxed{\vec{v} = \dot{\rho}\hat{\rho} + \rho\dot{\phi}\hat{\phi} + \dot{z}\hat{z}}$$

Acceleration:

$$\begin{aligned}\vec{a} &= \frac{d\vec{v}}{dt} \\ &= \ddot{\rho}\hat{\rho} + \dot{\rho}\frac{d\hat{\rho}}{dt} + \dot{\phi}\dot{\rho}\hat{\phi} + \ddot{\phi}\hat{\phi} + \rho\dot{\phi}\dot{\phi}\hat{\phi} + \rho\dot{\phi}\frac{d\hat{\phi}}{dt} + \ddot{z}\hat{z}\end{aligned}$$

$$\text{Now: } \frac{d\hat{\phi}}{dt} = \frac{\partial \hat{\phi}}{\partial \rho} \dot{\rho} + \frac{\partial \hat{\phi}}{\partial \phi} \dot{\phi} + \frac{\partial \hat{\phi}}{\partial z} \dot{z}$$

(2)

From App. A:

$$\frac{\partial \hat{\phi}}{\partial r} = 0, \quad \frac{\partial \hat{\phi}}{\partial \theta} = -\hat{r}, \quad \frac{\partial \hat{\phi}}{\partial z} = 0$$

Thus,  $\frac{d\hat{\phi}}{dt} = -\dot{\phi}\hat{r}$

$$\rightarrow \vec{a} = \ddot{r}\hat{r} + \dot{r}\phi\hat{\phi} + \dot{r}\dot{\phi}\hat{\phi} + \ddot{\rho}\dot{\phi}\hat{\phi}$$

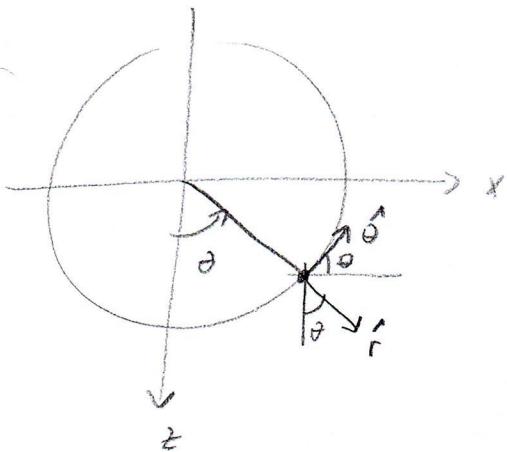
$$- \rho\dot{\phi}^2\hat{r} + \ddot{z}\hat{z}$$

$$= (\ddot{r} - \rho\dot{\phi}^2)\hat{r} + (2\dot{r}\dot{\phi} + \rho\ddot{\phi})\hat{\phi} + \ddot{z}\hat{z}$$

planar pendulum:

Prob 1.3

(1)



$$\begin{aligned}\vec{r} &= (\omega_0 \theta) \hat{z} + \sin \theta \hat{x} \\ \dot{\theta} &= -\sin \theta \hat{z} + (\omega_0 \theta) \hat{x}\end{aligned}$$

$$\frac{d\vec{r}}{dt} = \dot{\theta} [-\sin \theta \hat{z} + (\omega_0 \theta) \hat{x}] = \dot{\theta} \hat{r}$$

$$\frac{d\dot{\theta}}{dt} = \ddot{\theta} [-\omega_0 \theta \hat{z} - \sin \theta \hat{x}] = -\ddot{\theta} \hat{r}$$

$$\vec{r} = l \hat{r}$$

$$\vec{v} = \frac{d\vec{r}}{dt} = l \frac{d\theta}{dt} \hat{r} = l \dot{\theta} \hat{r} \quad (\text{tgt to circle})$$

$$\vec{a} = \frac{d\vec{v}}{dt} = l \ddot{\theta} \hat{r} + l \dot{\theta} \frac{d\hat{r}}{dt}$$

$$= \underbrace{l \ddot{\theta} \hat{r}}_{\text{tgt}} - \underbrace{l \dot{\theta}^2 \hat{r}}_{\text{cent. cent.}}$$

Net Force:

$$\vec{T} = -T \hat{r}$$

$$\vec{mg} = mg \cos \theta \hat{r} - mg \sin \theta \hat{z}$$

$$\vec{F} = (mg \cos \theta - T) \hat{r} - mg \sin \theta \hat{z}$$

$$\vec{ma} = ml \ddot{\theta} \hat{r} - ml \dot{\theta}^2 \hat{z}$$

$$\text{Thus, } \vec{F} = \vec{ma} \iff$$

$$\left. \begin{aligned} mg \cos \theta - T &= -ml \dot{\theta}^2 \\ -mg \sin \theta &= m l \dot{\theta} \end{aligned} \right\}$$

(2)

$$(a) T = mg \cos \theta + ml \dot{\theta}^2$$

$$(b) ml \ddot{\theta} = -mg \sin \theta$$

$$\ddot{\theta} = -\frac{g}{l} \sin \theta \quad [\text{oscillator equation}]$$

Multiply by  $\dot{\theta}$

$$\dot{\theta} \ddot{\theta} = -\frac{g}{l} \dot{\theta} \sin \theta$$

$$\frac{d}{dt} \left( \frac{1}{2} \dot{\theta}^2 \right) = \frac{d}{dt} \left( \frac{g}{l} \cos \theta \right)$$

$$\rightarrow \boxed{\frac{1}{2} \dot{\theta}^2 = \frac{g}{l} \cos \theta + C}$$

Rearrange:

$$\frac{1}{2} \dot{\theta}^2 - \frac{g}{l} \cos \theta = C$$

$$\frac{1}{2} ml^2 \dot{\theta}^2 - mg l \cos \theta = C ml^2$$

$$T + U = E \quad (\text{so } C \text{ is related to } E)$$

$$U = -mg l \cos \theta + mg l$$

$$= mg l (1 - \cos \theta)$$

$$\text{so } U = 0 \text{ at bottom.}$$

$$(c) T = mg \cos \theta + ml \dot{\theta}^2$$

max value of  $\theta$  having  $T \geq 0$  and  $\dot{\theta} = 0$

$$\rightarrow 0 = mg \cos \theta_0 \rightarrow \boxed{\theta_0 = \frac{\pi}{2}}$$

(d) minimum initial velocity for loop-the-loop?

Determine  $C$  in terms of initial velocity  $V_0 = l \dot{\theta}_0$

$$\frac{1}{2} \frac{V_0^2}{l^2} = \frac{g}{l} \cos \theta \Big|_{\theta=0} + C$$

at  $\theta=0$

(3)

$$\frac{1}{2} \frac{v_0^2}{l^2} = \frac{g}{l} + C$$

$$\rightarrow C = \frac{1}{2} \frac{v_0^2}{l^2} - \frac{g}{l}$$

$$\text{Thus, } \frac{1}{2} \dot{\theta}^2 - \frac{g}{l} \cos \theta = \frac{1}{2} \frac{v_0^2}{l^2} - \frac{g}{l}$$

$$\boxed{\frac{1}{2} \dot{\theta}^2 + \frac{g}{l} (1 - \cos \theta) = \frac{1}{2} \left( \frac{v_0}{l} \right)^2}$$

At top of loop-the-loop,  $\theta = \pi \rightarrow \cos \theta = -1$

$$T = mg \cos \theta + ml \dot{\theta}^2$$

$$\rightarrow O = -mg + ml \dot{\theta}^2$$

$$\text{so } \boxed{\dot{\theta} = \sqrt{\frac{g}{l}}} \quad \text{at top with } T=0.$$

Minimum  $v_0$ :

$$\text{At top } \frac{1}{2} \left( \sqrt{\frac{g}{l}} \right)^2 + \frac{g}{l} (1+1) = \frac{1}{2} \left( \frac{v_0}{l} \right)^2$$

$$\frac{1}{2} \left( \frac{g}{l} \right) + 2 \left( \frac{g}{l} \right) = \frac{1}{2} \left( \frac{v_0}{l} \right)^2$$

$$\frac{5}{2} \frac{g}{l} = \frac{1}{2} \left( \frac{v_0}{l} \right)^2$$

$$\rightarrow \boxed{v_0 = \sqrt{5gl}}$$

(4)

Simple conservation of energy argument:

$$\text{At bottom} \quad E = TE = \frac{1}{2}mv_0^2$$

$$\text{At top} \quad E = 2mgl + \frac{1}{2}mv^2$$

~~mg~~

$$= 2mgl + \frac{1}{2}mgl$$

$$= \frac{5}{2}mgl$$

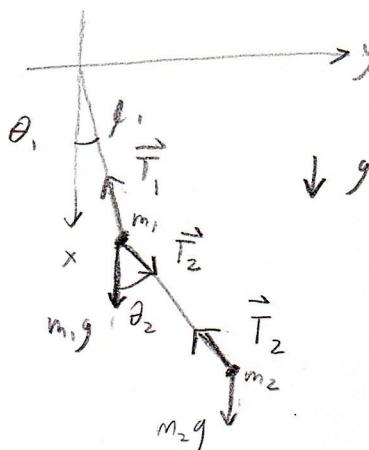
Centripetal force  
given only by  
 $mg \rightarrow$   
 $mg = \frac{mv^2}{R}$   
 $\rightarrow v^2 = gR$

$$\text{Thus, } \frac{1}{2}mv_0^2 = \frac{5}{2}mgl$$

$$v_0 = \sqrt{5gl}$$

# Double pendulum: Prob 1,4

①



$$\begin{aligned}\vec{F}_1 &= l_1 \cos \theta_1 \hat{x} + l_1 \sin \theta_1 \hat{y} \\ \vec{F}_2 &= \vec{F}_1 + l_2 \cos \theta_2 \hat{x} + l_2 \sin \theta_2 \hat{y} \\ &= (l_1 \cos \theta_1 + l_2 \cos \theta_2) \hat{x} \\ &\quad + (l_1 \sin \theta_1 + l_2 \sin \theta_2) \hat{y}\end{aligned}$$

calculate velocity and acceleration vectors

$$\vec{v}_1 = -l_1 \sin \theta_1 \dot{\theta}_1 \hat{x} + l_1 \cos \theta_1 \dot{\theta}_1 \hat{y}$$

$$= l_1 \dot{\theta}_1 [-\sin \theta_1 \hat{x} + \cos \theta_1 \hat{y}]$$

$$\vec{v}_2 = (-l_1 \sin \theta_1 \dot{\theta}_1 - l_2 \sin \theta_2 \dot{\theta}_2) \hat{x}$$

$$+ (l_1 \cos \theta_1 \dot{\theta}_1 + l_2 \cos \theta_2 \dot{\theta}_2) \hat{y}$$

$$= l_1 \dot{\theta}_1 [-\sin \theta_1 \hat{x} + \cos \theta_1 \hat{y}]$$

$$+ l_2 \dot{\theta}_2 [-\sin \theta_2 \hat{x} + \cos \theta_2 \hat{y}]$$

$$\vec{a}_1 = l_1 \ddot{\theta}_1 [-\sin \theta_1 \hat{x} + \cos \theta_1 \hat{y}]$$

$$+ l_1 \dot{\theta}_1 [-\cos \theta_1 \dot{\theta}_1 \hat{x} - \sin \theta_1 \dot{\theta}_1 \hat{y}]$$

$$= l_1 \ddot{\theta}_1 [-\sin \theta_1 \hat{x} + \cos \theta_1 \hat{y}] - l_1 \dot{\theta}_1^2 [\cos \theta_1 \hat{x} + \sin \theta_1 \hat{y}]$$

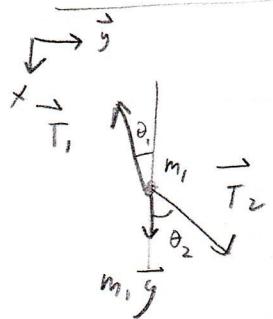
$$\vec{a}_2 = l_1 \ddot{\theta}_1 [-\sin \theta_1 \hat{x} + \cos \theta_1 \hat{y}] - l_1 \dot{\theta}_1^2 [\cos \theta_1 \hat{x} + \sin \theta_1 \hat{y}]$$

$$+ l_2 \ddot{\theta}_2 [-\sin \theta_2 \hat{x} + \cos \theta_2 \hat{y}] - l_2 \dot{\theta}_2^2 [\cos \theta_2 \hat{x} + \sin \theta_2 \hat{y}]$$

$$\vec{q}_1 = -l_1 \left( \ddot{\theta}_1 \sin \theta_1 + \dot{\theta}_1^2 \cos \theta_1 \right) \hat{x} + l_1 \left( \ddot{\theta}_1 \cos \theta_1 - \dot{\theta}_1^2 \sin \theta_1 \right) \hat{y} \quad (2)$$

$$\vec{q}_2 = \left[ -l_1 \left( \ddot{\theta}_1 \sin \theta_1 + \dot{\theta}_1^2 \cos \theta_1 \right) - l_2 \left( \ddot{\theta}_2 \sin \theta_2 + \dot{\theta}_2^2 \cos \theta_2 \right) \right] \hat{x}$$

$$+ \left[ l_1 \left( \ddot{\theta}_1 \cos \theta_1 - \dot{\theta}_1^2 \sin \theta_1 \right) + l_2 \left( \ddot{\theta}_2 \cos \theta_2 - \dot{\theta}_2^2 \sin \theta_2 \right) \right] \hat{y}$$

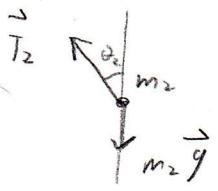


$$\vec{F}_1 = -T_1 \cos \theta_1 \hat{x} - T_1 \sin \theta_1 \hat{y}$$

$$+ T_2 \cos \theta_2 \hat{x} + T_2 \sin \theta_2 \hat{y} + m_1 g \hat{y}$$

$$= (m_1 g - T_1 \cos \theta_1 + T_2 \cos \theta_2) \hat{x}$$

$$+ (-T_1 \sin \theta_1 + T_2 \sin \theta_2) \hat{y}$$



$$\vec{F}_2 = -T_2 \cos \theta_2 \hat{x} - T_2 \sin \theta_2 \hat{y} + m_2 g \hat{x}$$

$$= (m_2 g - T_2 \cos \theta_2) \hat{x} - T_2 \sin \theta_2 \hat{y}$$

2nd law:

$$m_1 g - T_1 \cos \theta_1 + T_2 \cos \theta_2 = -m_1 l_1 \left( \ddot{\theta}_1 \sin \theta_1 + \dot{\theta}_1^2 \cos \theta_1 \right) \quad (1)$$

$$-T_1 \sin \theta_1 + T_2 \sin \theta_2 = m_1 l_1 \left( \ddot{\theta}_1 \cos \theta_1 - \dot{\theta}_1^2 \sin \theta_1 \right) \quad (2)$$

$$m_2 g - T_2 \cos \theta_2 = m_2 \left[ -l_1 \left( \ddot{\theta}_1 \sin \theta_1 + \dot{\theta}_1^2 \cos \theta_1 \right) - l_2 \left( \ddot{\theta}_2 \sin \theta_2 + \dot{\theta}_2^2 \cos \theta_2 \right) \right] \quad (3)$$

$$-T_2 \sin \theta_2 = m_2 \left[ l_1 \left( \ddot{\theta}_1 \cos \theta_1 - \dot{\theta}_1^2 \sin \theta_1 \right) + l_2 \left( \ddot{\theta}_2 \cos \theta_2 - \dot{\theta}_2^2 \sin \theta_2 \right) \right] \quad (4)$$

(3)

From (4)

$$\left[ \begin{array}{l} T_2 \sin \theta_2 = -m_2 [ l_1 (\ddot{\theta}_1 \cos \theta_1 - \dot{\theta}_1^2 \sin \theta_1) \\ \quad + l_2 (\ddot{\theta}_2 \cos \theta_2 - \dot{\theta}_2^2 \sin \theta_2) ] \end{array} \right]$$

Add (2) and (4):

$$\begin{aligned} -T_1 \sin \theta_1 &= [(m_1 + m_2) l_1 (\ddot{\theta}_1 \cos \theta_1 - \dot{\theta}_1^2 \sin \theta_1) \\ &\quad + m_2 l_2 (\ddot{\theta}_2 \cos \theta_2 - \dot{\theta}_2^2 \sin \theta_2)] \end{aligned}$$

so

$$\left[ \begin{array}{l} T_1 \sin \theta_1 = -(m_1 + m_2) l_1 (\ddot{\theta}_1 \cos \theta_1 - \dot{\theta}_1^2 \sin \theta_1) \\ \quad - m_2 l_2 (\ddot{\theta}_2 \cos \theta_2 - \dot{\theta}_2^2 \sin \theta_2) \end{array} \right]$$

(3)  $\sin \theta_2 - (4) \cos \theta_2 :$ 

$$\begin{aligned} m_2 g \sin \theta_2 &= -m_2 \sin \theta_2 \quad [ l_1 (\ddot{\theta}_1 \sin \theta_1 + \dot{\theta}_1^2 \cos \theta_1) \\ &\quad + l_2 (\ddot{\theta}_2 \sin \theta_2 + \dot{\theta}_2^2 \cos \theta_2) ] \end{aligned}$$

$$\begin{aligned} &- m_2 \cos \theta_2 [ l_1 (\ddot{\theta}_1 \cos \theta_1 - \dot{\theta}_1^2 \sin \theta_1) \\ &\quad + l_2 (\ddot{\theta}_2 \cos \theta_2 - \dot{\theta}_2^2 \sin \theta_2) ] \end{aligned}$$

$$\begin{aligned} &= -m_2 [ l_1 \ddot{\theta}_1 (\sin \theta_1 \sin \theta_2 + \cos \theta_1 \cos \theta_2) \\ &\quad + l_2 \ddot{\theta}_2 (\sin^2 \theta_2 + \cos^2 \theta_2) \\ &\quad + l_1 \dot{\theta}_1^2 (\cos \theta_1 \sin \theta_2 - \sin \theta_1 \cos \theta_2) ] \end{aligned}$$

(4)

Now,

$$\cos(A \pm B) = \cos A \cos B \mp \sin A \sin B$$

$$\sin(A \pm B) = \sin A \cos B \pm \cos A \sin B$$

$$m_2 g \sin \theta_2 = -m_2 [ l_1 \ddot{\theta}_1 \cos(\theta_1 - \theta_2) + l_2 \ddot{\theta}_2 \\ - l_1 \dot{\theta}_1^2 \sin(\theta_1 - \theta_2) ]$$

$$g \sin \theta_2 = -l_1 \ddot{\theta}_1 \cos \Delta + l_1 \dot{\theta}_1^2 \sin \Delta - l_2 \ddot{\theta}_2$$

$$(1) \sin \theta_1 - (2) \cos \theta_1 + (3) \sin \theta_1 - (4) \cos \theta_1 :$$

$$LHS = m_1 g \sin \theta_1 - T_1 \cos \theta_1 \sin \theta_1 + T_2 \cos \theta_2 \sin \theta_1 \\ + T_1 \sin \theta_1 \cos \theta_1 - T_2 \sin \theta_2 \cos \theta_1$$

$$+ m_2 g \sin \theta_1 - T_2 \cos \theta_2 \sin \theta_1 + T_2 \sin \theta_2 \cos \theta_1$$

$$= (m_1 + m_2) g \sin \theta_1$$

$$RHS = -m_1 l_1 \left[ \sin \theta_1 (\ddot{\theta}_1 \sin \theta_1 + \dot{\theta}_1^2 \cos \theta_1) \right. \\ \left. + \cos \theta_1 (\dot{\theta}_1 \cos \theta_1 - \dot{\theta}_1^2 \sin \theta_1) \right]$$

$$+ m_2 \left[ -l_1 \sin \theta_1 (\ddot{\theta}_1 \sin \theta_1 + \dot{\theta}_1^2 \cos \theta_1) \right.$$

$$\left. - l_1 \cos \theta_1 (\dot{\theta}_1 \cos \theta_1 - \dot{\theta}_1^2 \sin \theta_1) \right]$$

$$- l_2 \sin \theta_1 (\ddot{\theta}_2 \sin \theta_2 + \dot{\theta}_2^2 \cos \theta_2)$$

$$- l_2 \cos \theta_1 (\dot{\theta}_2 \cos \theta_2 - \dot{\theta}_2^2 \sin \theta_2) \Big]$$

(5)

$$R M M 2 \cancel{= l_1 l_2 \dot{\theta}_1 \dot{\theta}_2} = m_2 l_2 \dot{\theta}_2$$

$$\begin{aligned}
 R H S &= -m_1 l_1 \ddot{\theta}_1 (\sin^2 \theta_1 + \omega^2 \theta_1) \\
 &\quad - m_2 l_2 \ddot{\theta}_2 (\sin^2 \theta_2 + \omega^2 \theta_2) \\
 &\quad - m_2 l_2 \dot{\theta}_2^2 (\cos \theta_1 \cos \theta_2 + \sin \theta_1 \sin \theta_2) \\
 &\quad - m_2 l_2 \dot{\theta}_2^2 (\sin \theta_1 \cos \theta_2 - \cos \theta_1 \sin \theta_2) \\
 &= -(m_1 + m_2) l_1 \ddot{\theta}_1 - m_2 l_2 \dot{\theta}_2^2 \cos \Delta - m_2 l_2 \dot{\theta}_2^2 \sin \Delta
 \end{aligned}$$

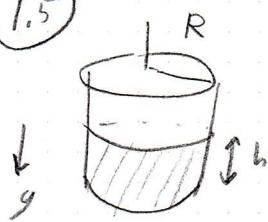
thus,

$$(m_1 + m_2) g \sin \theta_1 = -(m_1 + m_2) l_1 \ddot{\theta}_1 - m_2 l_2 (\dot{\theta}_2^2 \cos \Delta + \dot{\theta}_2^2 \sin \Delta)$$

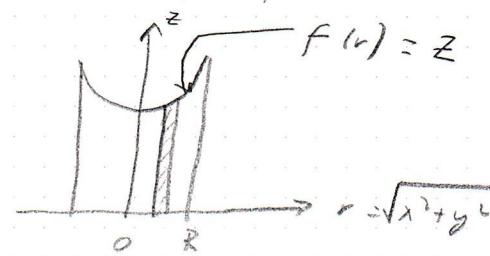
Problem: Shape of surface of water in a rotating bucket

①

(1.5)



Azimuthally symmetric:



~~Hydrostatic PE~~

$$U_{tot} = \text{rotational PE} + U_{gravity}$$

$U_{centrifugal}$

$$\begin{aligned} \vec{F}_{\text{centrifugal}} &= -m \vec{\omega} \times (\vec{\omega} \times \vec{r}) \\ &= m \omega^2 r \hat{r} \end{aligned}$$

$$\rightarrow U_{\text{centrifugal}} = -\frac{1}{2} m \omega^2 r^2$$

Thus,

$$\begin{aligned} dU_{tot} &= -\frac{1}{2} dm \omega^2 r^2 + dm \cdot g \cdot \underbrace{\frac{1}{2} z(r)}_{\text{avg height}} \\ &= \frac{1}{2} [\rho \cancel{\pi} r z dr \omega^2 r^2 + \rho \cancel{\pi} r z dr g z] \\ &= \pi \rho [-\omega^2 r^3 z + g r z^2] dr \end{aligned}$$

Constraint: Volume =  $\pi R^2 h$

$$\int_0^R z \pi r dr z = \pi R^2 h$$

so 
$$\int_0^R 2 \pi r z dr - \pi R^2 h = 0$$

Extremize:  $I[z] = \int_0^R \pi \rho [-\omega^2 r^3 z + g r z^2] dr$

subject to  $J[z] = 2\pi \int r z dr - \pi R^2 h = 0$

$$F(z, z', r) = \pi \rho [E \omega^2 r^3 z + g r - z^2]$$

$$G(z, z', r) = 2\pi r z$$

$$\text{i)} \left( \frac{d}{dr} \left( \frac{\partial F}{\partial z'} \right)^T - \frac{\partial F}{\partial z} \right) + \lambda \left[ \frac{d}{dr} \left( \frac{\partial G}{\partial z'} \right)^T - \frac{\partial G}{\partial z} \right] = 0$$

$$\text{ii)} \int_0^R 2\pi r z \, dr = \pi R^2 h$$

$$0 = \frac{\partial F}{\partial z} + \lambda \frac{\partial G}{\partial z}$$

$$= \pi \rho [E \omega^2 r^3 + 2grz] + \lambda 2\pi r$$

$$= -\pi \rho \omega^2 r^3 + 2\pi \rho grz + \lambda 2\pi r$$

$$= -\rho \omega^2 r^3 + 2\rho grz + 2\lambda r$$

$$\rightarrow Z = \frac{-2\lambda r + \rho \omega^2 r^3}{2\rho gr}$$

$$= \frac{1}{2\rho g} [Ez\lambda + \rho \omega^2 r^2]$$

$$= \frac{-\lambda}{\rho g} + \frac{1}{2g} \omega^2 r^2 = A + Br^2$$

$$\text{Substitute: } \pi R^2 h = \int_0^R 2\pi r \left[ \frac{-\lambda}{\rho g} + \frac{1}{2g} \omega^2 r^2 \right] dr$$

$$= -\frac{2\pi \lambda}{\rho g} \frac{r^2}{2} \Big|_0^R + \frac{\pi \omega^2}{g} \frac{r^4}{4} \Big|_0^R$$

$$= -\frac{\pi \lambda}{\rho g} R^2 + \frac{\pi}{4g} \omega^2 R^4$$

(3)

$$DR^2 h = \pi R^2 \left[ -\frac{1}{\rho g} + \frac{R^2 \omega^2}{4g} \right]$$

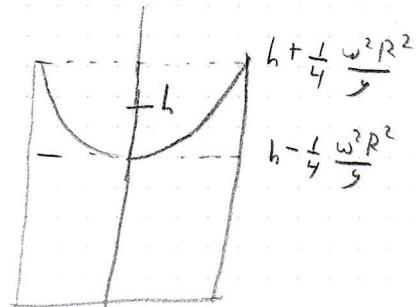
$$h = \frac{-1}{\rho g} + \frac{R^2 \omega^2}{4g}$$

$$\rightarrow \boxed{\lambda} = \rho g \left( h + \frac{R^2 \omega^2}{4g} \right)$$

$$= -\rho g h + \frac{1}{4} \rho \omega^2 R^2$$

Thus,

$$\begin{aligned} \boxed{z} &= -\frac{1}{\rho g} + \frac{1}{2g} \omega^2 r^2 \\ &= h - \frac{R^2 \omega^2}{4g} + \frac{1}{2g} \omega^2 r^2 \\ &= h + \frac{\omega^2}{2g} \left[ r^2 - \frac{R^2}{2} \right] \end{aligned}$$



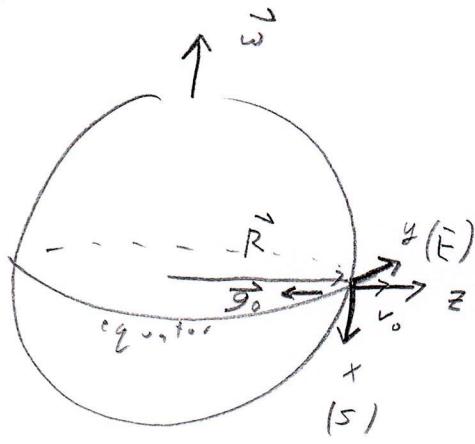
NOTE:  $r=0 \rightarrow z(0) = h - \frac{1}{4} \frac{\omega^2 R^2}{g}$

$r=R \quad z(R) = h + \frac{1}{4} \frac{\omega^2 R^2}{g}$

chart volume:

$$\begin{aligned} 2\pi \int_0^R r z dr &= 2\pi \int_0^R r \left( h + \frac{\omega^2}{2g} \left[ r^2 - \frac{R^2}{2} \right] \right) dr \\ &= 2\pi \left[ \int_0^R h r dr + \frac{\omega^2}{2g} \int_0^R \left( r^3 - \frac{R^2 r}{2} \right) dr \right] \\ &= 2\pi \left[ h \frac{R^2}{2} + \frac{\omega^2}{2g} \left( \frac{R^4}{4} - \frac{R^2 R^2}{2} \right) \right] \\ &= \pi R^2 h \quad \checkmark \end{aligned}$$

Prob: (1.6) Launch from equator (Q)



$$m \vec{a} = \vec{F} - m \vec{\omega} \times \vec{v} - 2m \vec{\omega} \times \vec{v}$$

$$- m \vec{\omega} \times (\vec{\omega} \times \vec{r}) - m R_f \ddot{\vec{r}}$$

$$\vec{R}_f = \vec{\omega} \times (\vec{\omega} \times \vec{R})$$

$$\dot{\vec{\omega}} = 0$$

$$\vec{F} = m \vec{g}_0$$

$$\vec{F} = m \vec{g}_0$$

$$\vec{\omega} = \text{const}$$

a) Thus,

$$m \vec{a} = m \left[ \vec{g}_0 - \vec{\omega} \times (\vec{\omega} \times (\vec{r} + \vec{R})) \right] - 2m \vec{\omega} \times \vec{v}$$

$$\boxed{\vec{a} = \vec{g}_0 - \vec{\omega} \times (\vec{\omega} \times \vec{r}) - \vec{\omega} \times (\vec{\omega} \times \vec{R}) - 2\vec{\omega} \times \vec{v}}$$

Let  $\vec{r} = x \hat{x} + y \hat{y} + z \hat{z}$  (wrt non-inertial frame)

$$\vec{v} = \dot{x} \hat{x} + \dot{y} \hat{y} + \dot{z} \hat{z}$$
 time derivative wrt non-inertial

$$\vec{a} = \ddot{x} \hat{x} + \ddot{y} \hat{y} + \ddot{z} \hat{z}$$
 frame

Now,

$$\vec{g}_0 = -g_0 \hat{z}$$

$$\vec{R} = R \hat{z}$$

$$\vec{\omega} = -\omega \hat{x}$$

Thus,

$$\vec{\omega} \times \vec{r} = -\omega \hat{x} \times (x \hat{x} + y \hat{y} + z \hat{z})$$

$$= -wy \hat{z} + wz \hat{y}$$

$$\vec{\omega} \times (\vec{\omega} \times \vec{r}) = -\omega \hat{x} \times (-wy \hat{z} + wz \hat{y})$$

$$= -w^2 y \hat{y} - w^2 z \hat{z}$$

(2)

$$\vec{\omega} \times \vec{R} = -\omega \hat{x} \times R \hat{z}$$

$$= \omega R \hat{y}$$

$$\vec{\omega} \times (\vec{\omega} \times \vec{R}) = -\omega \hat{x} \times \omega R \hat{y}$$

$$= -\omega^2 R \hat{z}$$

$$\vec{\omega} \times \vec{v} = -\omega \hat{x} \times (\hat{x} \hat{x} + \hat{y} \hat{y} + \hat{z} \hat{z})$$

$$= -\omega \hat{y} \hat{z} + \omega \hat{z} \hat{y}$$

$$\vec{r} = \vec{g}_o - \vec{\omega} \times (\vec{\omega} \times \vec{r}) - \vec{\omega} \times (\vec{\omega} \times \vec{R}) - 2\vec{\omega} \times \vec{v}$$

$$\ddot{x} \hat{x} + \ddot{y} \hat{y} + \ddot{z} \hat{z} = -g_o \hat{z} + \omega^2 y \hat{y} + \omega^2 z \hat{z} + \omega^2 R \hat{z}$$

$$+ 2\omega \dot{y} \hat{z} - 2\omega \dot{z} \hat{y}$$

$$= \hat{y} (\omega^2 y - 2\omega \dot{z})$$

$$+ \hat{z} (-g_o + \omega^2 z + \omega^2 R + 2\omega \dot{y})$$

Thus,

$\ddot{x} = 0$	}
$\ddot{y} = \omega^2 y - 2\omega \dot{z}$	
$\ddot{z} = (-g_o + \omega^2 R) + \omega^2 z + 2\omega \dot{y}$	

(3)

b) If  $\omega = 0$ 

$$\ddot{x} = 0, \quad \ddot{y} = 0, \quad \ddot{z} = -g_0$$

$$\rightarrow y_0(t) = 0, \quad z_0(t) = -\frac{1}{2}g_0 t^2 + v_0 t \quad (z_0(0) = 0)$$

Perturbed trajectory

$$y(t) = \psi(t), \quad z(t) = z_0(t) + \xi(t)$$

To 1<sup>st</sup> order:

$$\boxed{\ddot{\psi} = \cancel{\omega^2 \ddot{y}} - 2\omega (\dot{z}_0(t) + \dot{\xi}(t))}$$

$$= -2\omega \dot{z}_0$$

$$= -2\omega (v_0 - g_0 t)$$

$$\ddot{z}_0 + \ddot{\xi} = (-g_0 + \omega^2 R) + \omega^2 z + 2\omega \psi$$

$\uparrow$                        $\uparrow$                        $\cup$   
 2<sup>nd</sup> order          2<sup>nd</sup> order          2<sup>nd</sup> order

$$\rightarrow -g_0 + \ddot{\xi} = -g_0$$

$$\text{or } \boxed{\ddot{\xi} = 0}$$

c) Solve these equations

$$\ddot{\xi} = 0 \rightarrow \xi(t) = At + B$$

But  $A = 0, B = 0$  in order for  $z(t) = z_0(t) + \xi(t)$   
 to satisfy  $z(0) = 0, \dot{z}(0) = v_0$ , Then  $\boxed{\xi(t) = 0}$

$$\ddot{\psi} = -2\omega(v_0 - g_0 t)$$

$$= -2\omega v_0 + 2\omega g_0 t$$

(4)

$$\rightarrow \psi = -2\omega v_0 t + \omega g_0 t^2 + A$$

$$\rightarrow \psi = -\omega v_0 t^2 + \frac{1}{3} \omega g_0 t^3 + At + B$$

Again, in order for  $y(t) = \psi(t)$  to satisfy  
 $y(0) = 0, \dot{y}(0) = 0 \rightarrow A = 0, B = 0$ .

Thus,  $\boxed{\psi(t) = \frac{1}{3} \omega g_0 t^3 - \omega v_0 t^2}$ ,

- d) Suppose the project is launched to a height  $h = 100 \text{ ft}_m$ .  
 How far from the launch site does it land?

Vertical motion: At top of trajectory  $\dot{z}(T_{\frac{1}{2}}) = 0$

$$0 = -g_0 T_{\frac{1}{2}} + v_0$$

$$\boxed{T_{\frac{1}{2}} = \frac{v_0}{g_0}}$$

$$\text{Time back to Earth: } T = 2 T_{\frac{1}{2}} = \boxed{\frac{2 v_0}{g_0}}$$

At top of trajectory

$$\begin{aligned} h = z_0(T_{\frac{1}{2}}) &= v_0 T_{\frac{1}{2}} - \frac{1}{2} g_0 (T_{\frac{1}{2}})^2 \\ &= \frac{v_0^2}{g_0} - \frac{1}{2} g_0 \left( \frac{v_0^2}{g_0^2} \right) \\ &= \frac{1}{2} \frac{v_0^2}{g_0} \end{aligned}$$

Thus,  $\boxed{v_0 = \sqrt{2 g_0 h}}$  — initial velocity required  
 to reach height  $h$

Distance from landing site:

$$\begin{aligned}
 \psi(T) &= \frac{1}{3} \omega g_0 T^3 - \omega v_0 T^2 \\
 &= \frac{1}{3} \omega g_0 \left(\frac{2v_0}{g_0}\right)^3 - \omega v_0 \left(\frac{2v_0}{g_0}\right)^2 \\
 &= \frac{8}{3} \frac{\omega v_0^3}{g_0^2} - 4 \frac{\omega v_0^3}{g_0^2} \\
 &= \left( -\frac{4}{3} \frac{\omega v_0^3}{g_0^2} \right)
 \end{aligned}$$

| add to this w, +

$$\begin{aligned}
 \psi(T) &= -\frac{4}{3} \frac{\omega}{g_0^2} \overbrace{\left(2g_0 h\right)^{3/2}}^{\text{velocity}^3 \cdot \frac{1}{T}} \\
 &= -\frac{4}{3} \frac{\omega}{g_0^{1/2}} 2\sqrt{2} h^{3/2} \\
 &= -\frac{8\sqrt{2}}{3} \frac{\omega h}{g_0} \sqrt{\frac{h}{g_0}} \\
 &\quad \begin{array}{l} \text{velocity} \\ \text{time} \end{array} \\
 &= \frac{\left(\frac{L}{T}\right)^3 \cdot \frac{1}{T}}{\frac{L^2}{T^4}} = L
 \end{aligned}$$

Ans

for  $t_{\text{max}}$

Substitute numbers:

$$\begin{aligned}
 \psi(T) &= -\frac{8\sqrt{2}}{3} \left( \frac{2\pi}{24.3600} \right) \left( 100 \times 10^3 \text{ m} \right) \sqrt{\frac{100 \times 10^3 \text{ m}}{9.8 \text{ m/s}^2}} \\
 &= \boxed{[-2.77 \text{ fm}]}
 \end{aligned}$$

Prob(1.7) Droped from equator

①

Following the analysis in the previous problem:

$$y_0(t) = 0, \quad z_0(t) = h - \frac{1}{2}g_0 t^2$$

$$y(t) = \psi(t), \quad z(t) = z_0(t) + s(t)$$

$$\dot{s} = 0 \rightarrow s(t) = At + B$$

In order for  $z(t) = z_0(t) + s(t)$  to satisfy

$$z(0) = h, \quad \dot{z}(0) = 0 \rightarrow A = 0, B = 0$$

$$\ddot{\psi} = -2\omega \dot{z}_0 = 2\omega g_0 t$$

$$\rightarrow \psi(t) = \frac{1}{3}\omega g_0 t^3 + At + B$$

In order for  $y(t) = \psi(t)$  to satisfy

$$y(0) = 0, \quad \dot{y}(0) = 0 \rightarrow A = 0, B = 0$$

Thus,

$$\left. \begin{aligned} z(t) &= h - \frac{1}{2}g_0 t^2 \\ y(t) &= \frac{1}{3}\omega g_0 t^3 \end{aligned} \right\}$$

object hits the ground at time  $T$ :

$$0 = h - \frac{1}{2}g_0 T^2 \rightarrow T = \sqrt{\frac{2h}{g_0}}$$

Distance traveled in  $y$ -direction

$$y(T) = \frac{1}{3}\omega g_0 \left(\frac{2h}{g_0}\right)^{3/2} = \frac{1}{3}2\sqrt{2}\omega g_0 \frac{h^{3/2}}{g_0^{3/2}}$$

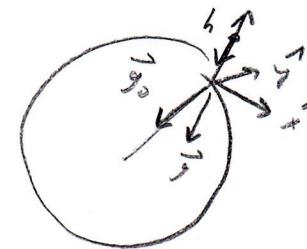
$$= \boxed{\frac{2\sqrt{2}}{3} \omega h \sqrt{\frac{h}{g_0}}} = \boxed{0.69 \text{ fm}} \quad (\text{to the east})$$

Prob: Drop from arbitrary latitude

Again define  $\vec{z}$  w.t  $\vec{g} = \vec{g}_0 - \vec{\omega} \times (\vec{\omega} \times \vec{R})$

Equations:  $\vec{\omega} \approx -\omega \sin \theta \vec{x} + \omega \cos \theta \vec{z}$

$$\ddot{x} = \omega \cos \theta (\omega \sin \theta z + \omega \cos \theta x) + 2 \omega \cos \theta y$$



$$\ddot{y} = \omega^2 y - 2(\omega \sin \theta \dot{z} + \omega \cos \theta \dot{x})$$

$$\ddot{z} = -g + 2\omega \sin \theta \dot{y} + \omega \sin \theta (\omega \sin \theta z + \omega \cos \theta x)$$

0<sup>th</sup> order:

$$\ddot{z} = -g$$

$$z(0) = h, \quad \dot{z}(0) = 0$$

$$\ddot{y} = 0$$

$$\ddot{z} = 0$$

$$\boxed{z(t) = h - \frac{1}{2}gt^2} \rightarrow z(0) = h$$

$$\dot{z}(t) = -gt \rightarrow \dot{z}(0) = 0$$

$$\ddot{z}(t) = -g$$

1<sup>st</sup> order:

$$\ddot{y} = -2\omega \sin \theta z$$

$$= +2\omega \sin \theta gt$$

$$\rightarrow \ddot{y} = \omega \sin \theta gt^2$$

$$\rightarrow \boxed{y = \frac{1}{3}\omega \sin \theta g t^3}$$

2<sup>nd</sup> order:

$$\ddot{x} = \omega^2 \sin \theta \cos \theta [h + \frac{1}{2}gt^2] + \frac{\omega^2 \cos^2 \theta}{\cos \theta} x$$

$$+ 2\omega \cos \theta [\omega \sin \theta gt^2]$$

2<sup>nd</sup> order

$\downarrow z$

$\uparrow y$

(2)

$$\begin{aligned}
 x' &= \omega^2 \sin \theta \cos \theta [h - \frac{1}{2}gt^2] + 2\omega^2 \sin \theta \cos \theta gt^2 \\
 &= \omega^2 \sin \theta \cos \theta [h - \frac{1}{2}gt^2 + 2gt^2] \\
 &= \omega^2 \sin \theta \cos \theta [h + \frac{3}{2}gt^2]
 \end{aligned}$$

$$\rightarrow x' = \omega^2 \sin \theta \cos \theta [ht + \frac{1}{2}gt^3]$$

$$\rightarrow \boxed{x = \omega^2 \sin \theta \cos \theta [\frac{1}{2}ht^2 + \frac{1}{8}gt^4]}$$

Displacement: T given by  $z/T = 0 = h - \frac{1}{2}gt^2 \rightarrow T = \sqrt{\frac{2h}{g}}$

$$\Delta y = y(T)$$

$$= \frac{1}{3} \omega \sin \theta g \left( \frac{2h}{g} \right)^{3/2}$$

$$= \frac{2^{3/2}}{3} \omega \sin \theta \frac{h^{3/2}}{g^{1/2}}$$

$$= \frac{2\sqrt{2}}{3} \omega \sin \theta \sqrt{\frac{h^3}{g}}$$

$$= \boxed{\frac{2}{3} \omega \sin \theta \sqrt{\frac{2h^3}{g}}}$$

For Launch A

$$\text{vs. } -\frac{8}{3} \omega \sin \theta \sqrt{\frac{2h^3}{g}} \quad (\text{differ by } -4x)$$

$$\Delta x = x(T)$$

$$= \omega^2 \sin \theta \cos \theta \left[ \frac{1}{2}h \left( \frac{2h}{g} \right) + \frac{1}{8}g \frac{4h^2}{g^2} \right]$$

$$= \omega^2 \sin \theta \cos \theta \left[ \frac{3}{2} \frac{h^2}{g} \right]$$

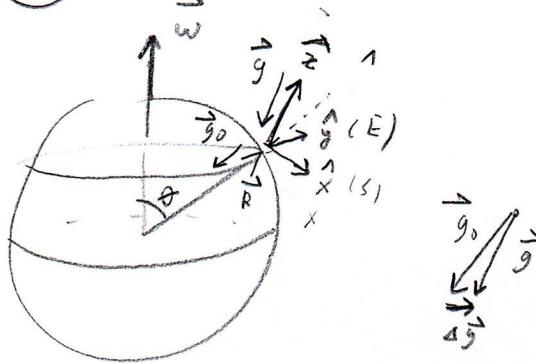
$$= \boxed{\frac{3}{2} \omega^2 \sin \theta \cos \theta \frac{h^2}{g}}$$

for launch A

$$\text{vs. } -8 \omega^2 \sin \theta \cos \theta \frac{h^2}{g}$$

$$\left( -\frac{16}{3} x \right)$$

Prob (1.8) Launch from arbitrary latitude ①



NOTE:  $\hat{z}$  is opposite  $\vec{g} \equiv \vec{g}_0 = \vec{\omega} \times (\vec{\omega} \times \vec{r})$

$$\vec{g} = -\vec{g} \hat{z}$$

$$\vec{v}_0 = v_0 \hat{z}$$

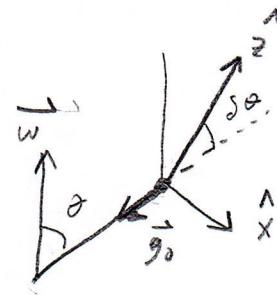
$$\vec{a} = (\vec{g}_0 - \vec{\omega} \times (\vec{\omega} \times \vec{r})) - \vec{\omega} \times (\vec{\omega} \times \vec{r}) - 2\vec{\omega} \times \vec{v}$$

$$= \vec{g}_0 - \vec{\omega} \times (\vec{\omega} \times \vec{r}) - 2\vec{\omega} \times \vec{v}$$

$$\vec{r} = \hat{x}\hat{x} + \hat{y}\hat{y} + \hat{z}\hat{z}$$

$$\vec{v} = \hat{x}\hat{x} + \hat{y}\hat{y} + \hat{z}\hat{z}$$

$$\vec{a} = \hat{x}\hat{x} + \hat{y}\hat{y} + \hat{z}\hat{z}$$



$$\vec{\omega} = -w \sin(\theta - \delta\theta) \hat{x} + w \cos(\theta - \delta\theta) \hat{z}$$

$$\approx -w \sin \theta \hat{x} + w \cos \theta \hat{z}$$

$$\delta\theta = O(w^2)$$

$$[2^{\text{nd}}\text{-order}, \propto w^2]$$

Thus,  $\vec{\omega} \times \vec{R} = \cancel{(w \sin \theta \hat{x} + w \cos \theta \hat{z}) \times R \hat{z}}$

$$= R w \sin \theta \hat{y}$$

$$\cancel{(\vec{\omega} \times \vec{\omega} \times \vec{r})} = \cancel{(w \sin \theta \hat{x} + w \cos \theta \hat{z}) \times R w \sin \theta \hat{y}}$$

$$= \cancel{R w^2 \sin^2 \theta \hat{z}} = R w^2 \sin \theta \cos \theta \hat{x}$$

$$\vec{\omega} \times \vec{r} \approx (-w \sin \theta \hat{x} + w \cos \theta \hat{z}) \times (\hat{x}\hat{x} + \hat{y}\hat{y} + \hat{z}\hat{z})$$

$$= -w \sin \theta \hat{y} \hat{z} + w \sin \theta \hat{z} \hat{y}$$

$$+ w \cos \theta \hat{x} \hat{y} = w \cos \theta \hat{y} \hat{x}$$

$$= -w \cos \theta \hat{y} \hat{x} + (w \sin \theta \hat{z} + w \cos \theta \hat{x}) \hat{y}$$

$$- w \sin \theta \hat{y} \hat{z}$$

$$\vec{\omega}_x (\vec{\omega}_x \vec{r}) = \left( -\omega_{in} \theta \hat{x} + \omega \cos \theta \hat{z} \right) \times \left[ -\omega \cos \theta y \hat{x} + (\omega_{in} \theta z + \omega \cos \theta x) \hat{y} - \omega \sin \theta y \hat{z} \right] \quad (2)$$

$$= -\omega \sin \theta (\omega_{in} \theta z + \omega \cos \theta x) \hat{z} \\ - \omega^2 \sin^2 \theta y \hat{y} \\ - \omega^2 \cos^2 \theta y \hat{y} \\ - \omega \cos \theta (\omega_{in} \theta z + \omega \cos \theta x) \hat{x}$$

$$= -\omega \cos \theta (\omega_{in} \theta z + \omega \cos \theta x) \hat{x} \\ - \omega^2 y \hat{y} \\ - \omega \sin \theta (\omega_{in} \theta z + \omega \cos \theta x) \hat{z}$$

$$\vec{\omega} \times \vec{v} = (-\omega_{in} \theta \hat{x} + \omega \cos \theta \hat{z}) \times (\dot{x} \hat{x} + \dot{y} \hat{y} + \dot{z} \hat{z}) \\ = -\omega \cos \theta \dot{y} \hat{x} + (\omega_{in} \theta \dot{z} + \omega \cos \theta \dot{x}) \hat{y} - \omega \sin \theta \dot{y} \hat{z}$$

$$\ddot{x} \hat{x} + \ddot{y} \hat{y} + \ddot{z} \hat{z} = -g \hat{z} + \underbrace{\omega \cos \theta (\omega_{in} \theta z + \omega \cos \theta x)}_{+ \omega^2 y \hat{y}} \hat{x} \\ + \underbrace{\omega \sin \theta (\omega_{in} \theta z + \omega \cos \theta x)}_{+ R_0^2 \sin^2 \theta \cos \theta \hat{z} + R_0^2 \sin^2 \theta \cos \theta \hat{x}} \hat{z} \\ + 2 \omega \cos \theta \dot{y} \hat{x} \\ - \underbrace{2 (\omega_{in} \theta \dot{z} + \omega \cos \theta \dot{x})}_{+ 2 \omega \sin \theta \dot{y}} \hat{y} \\ + 2 \omega \sin \theta \dot{y} \hat{z}$$

(3)

$$\begin{aligned}
 &= \hat{x} \left[ \omega_{\text{rot}} \theta / (\omega_{\text{rot}} z + \omega_{\text{rot}} \theta x) \right. \\
 &\quad \left. + \cancel{\frac{\omega^2}{R} \sin \theta \cos \theta} + 2 \omega_{\text{rot}} \theta y \right] \\
 &+ \hat{y} \left[ \omega^2 y - 2 (\omega_{\text{rot}} \theta z + \omega_{\text{rot}} \theta x) \right] \\
 &+ \hat{z} \left[ -g + \cancel{\frac{\omega^2 \sin \theta}{R}} + 2 \omega_{\text{rot}} \theta y \right. \\
 &\quad \left. + \omega_{\text{rot}} \theta (\omega_{\text{rot}} z + \omega_{\text{rot}} \theta x) \right]
 \end{aligned}$$

$$\text{so } \ddot{x} = \omega_{\text{rot}} \theta (\omega_{\text{rot}} z + \omega_{\text{rot}} \theta x) + \cancel{\frac{\omega^2 \sin \theta \cos \theta}{R}} \\
 \quad + 2 \omega_{\text{rot}} \theta y$$

$$\ddot{y} = \omega^2 y - 2 (\omega_{\text{rot}} \theta z + \omega_{\text{rot}} \theta x)$$

$$\ddot{z} = \left( -g + \cancel{\frac{\omega^2 \sin \theta}{R}} \right) + 2 \omega_{\text{rot}} \theta y + \omega_{\text{rot}} \theta (\omega_{\text{rot}} z + \omega_{\text{rot}} \theta x)$$

0<sup>th</sup> order:  $\ddot{x}_0 = 0, \ddot{y}_0 = 0, \ddot{z}_0 = -g_0$

$z_0(t) = v_0 t - \frac{1}{2} g t^2$	using $z(0) = 0$
$y_0(t) = 0$	$z(0) = v_0$
$x_0(t) = 0$	using $x(0) = 0, \dot{x}(0) = 0$
$y(0) = 0, \dot{y}(0) = 0$	$y(0) = 0, \dot{y}(0) = 0$

1<sup>st</sup> order:  $\dot{x} = 0, \dot{y} = -2 \omega_{\text{rot}} \theta z, \dot{z} = -g$

Thus,  $\ddot{y} = -2 \omega_{\text{rot}} \theta (v_0 - g_0 t)$

$$\ddot{y} = -2 \omega_{\text{rot}} \theta v_0 + 2 \omega_{\text{rot}} \theta g_0 t$$

$$\rightarrow \dot{y} = -2 \omega_{\text{rot}} \theta v_0 t + \omega_{\text{rot}} \theta g_0 t^2$$

$$\rightarrow y = -\omega_{\text{rot}} \theta v_0 t^2 + \frac{1}{3} \omega_{\text{rot}} \theta g_0 t^3 \quad (1^{\text{st}} \text{ order})$$

Have to go to 2<sup>nd</sup> order to find  $x$ :

$$x'' = \underbrace{\omega_{\text{rot}}^2 \theta \left( \omega_{\text{lin}} \theta z + \omega_{\text{rot}} \theta x \right)}_{L_{1,1}^{(1)}} + \cancel{R\omega^2 \sin \theta \cos \theta} + 2\omega_{\text{rot}} \theta y'$$

$$\approx \omega^2 \sin \theta \cos \theta z + \cancel{R\omega^2 \sin \theta \cos \theta} + 2\omega_{\text{rot}} \theta y'$$

$$= \omega^2 \sin \theta \cos \theta \left[ \cancel{R} + v_0 t - \frac{1}{2} g_0 t^2 \right]$$

$$+ 2\omega_{\text{rot}} \theta \left[ -2\omega_{\text{lin}} \theta v_0 t + \omega_{\text{lin}} \theta g t^2 \right]$$

$$= \cancel{R\omega^2 \sin \theta \cos \theta} + \omega^2 \sin \theta \cos \theta v_0 t - \frac{1}{2} \omega^2 \sin \theta \cos \theta g t^2$$

$$- 4\omega^2 \sin \theta \cos \theta v_0 t + 2\omega^2 \sin \theta \cos \theta g t^2$$

$$= \cancel{R\omega^2 \sin \theta \cos \theta} - 3\omega^2 \sin \theta \cos \theta v_0 t + \frac{3}{2} \omega^2 \sin \theta \cos \theta g t^2$$

$$= \omega^2 \sin \theta \cos \theta \left[ \cancel{R} - 3v_0 t + \frac{3}{2} gt^2 \right]$$

$$= \omega^2 \sin \theta \cos \theta \left[ \cancel{R} - 3z(t) \right]$$

$$\rightarrow x' = \omega^2 \sin \theta \cos \theta \left[ \cancel{R} - \frac{3}{2} v_0 t^2 + \frac{1}{2} g t^3 \right] \quad \text{2nd order}$$

$$\rightarrow x = \omega^2 \sin \theta \cos \theta \left[ \cancel{R} - \frac{1}{2} v_0 t^3 + \frac{1}{8} g t^4 \right]$$

~~Time to fall~~

$$\text{Time of flight: } T = \frac{2v_0}{g} = \frac{2}{g} \sqrt{2g h} = \boxed{2 \sqrt{\frac{2h}{g}}} \quad \text{0th order.}$$



(57)

$$\Delta y = y(\tau)$$

$$= -\omega_{\text{rot}} \theta \cdot v_0 \left( \frac{2v_0}{g} \right)^2 + \frac{1}{2} \omega_{\text{rot}} \theta g \left( \frac{2v_0}{g} \right)^3$$

$$= -4 \omega_{\text{rot}} \theta \frac{v_0^3}{g^2} + \frac{8}{3} \omega_{\text{rot}} \theta g \frac{v_0^3}{g^3}$$

$$= -\frac{4}{3} \omega_{\text{rot}} \theta \frac{v_0^3}{g^2}$$

$$= -\frac{4}{3} \omega_{\text{rot}} \theta \frac{1}{g^2} (2gh)^{3/2} \quad \text{using } v_0 = \sqrt{2gh}$$

$$= \boxed{-\frac{8}{3} \omega_{\text{rot}} \theta \sqrt{\frac{2h^3}{g}}}$$

$$\Delta x = x(\tau)$$

$$= \omega^2 \sin \theta \cos \theta \left[ \underbrace{\frac{1}{2} R \left( \frac{2v_0}{g} \right)^2}_{-\frac{2Rv_0^2}{g^3}} - \underbrace{\frac{1}{2} v_0 \left( \frac{2v_0}{g} \right)^3}_{= 4 \frac{v_0^4}{g^3}} + \underbrace{\frac{1}{2} g \left( \frac{2v_0}{g} \right)^4}_{+ 2 \frac{v_0^4}{g^3}} \right]$$

$$= \omega^2 \sin \theta \cos \theta \left[ \cancel{\frac{Rv_0^2}{g^2}} - 2 \frac{v_0^4}{g^3} \right]$$

$$= \omega^2 \sin \theta \cos \theta \left[ \cancel{\frac{Rv_0^2}{g^2}} - 2 \left( \frac{4g^2h^2}{g^2} \right) \right]$$

$$= \omega^2 \frac{\sin(2\theta)}{2} \left[ \cancel{\frac{Rv_0^2}{g^2}} - 8 \left( \frac{h^2}{g} \right) \right]$$

$$= \boxed{-4 \omega^2 \sin(2\theta) \frac{h^2}{g}} = -8 \omega^2 \sin \theta \cos \theta \frac{h^2}{g}$$

$h = 100 \text{ km}$   
 $R = 6400 \text{ km}$

(6)

(c) Numerical example

$$26^\circ \text{ north latitude} \rightarrow \Theta = 90^\circ - 26^\circ = 64^\circ$$

$$h = 100 \text{ fm} = 10^5 \text{ m}$$

$$D = \sqrt{dx^2 + dy^2}$$

$$\text{where } dx = -8\omega^2 \sin \theta \cos \Theta \frac{h^2}{g}$$

$$\text{with } \omega = \frac{2\pi}{T \text{ day}}$$

$$g \approx 9.8 \text{ m/s}^2 \quad \left( \text{since } \frac{|f_g|}{g_0} \leq 0.003 = 0.3\% \text{ at equator} \right)$$

$$dy = -\frac{8}{3} \sqrt{\frac{2h^3}{g}} \omega \sin \theta$$

$$\rightarrow \boxed{\begin{aligned} dx &= -17 \text{ m} \\ dy &= -2.49 \times 10^3 \text{ m} \end{aligned}}$$

$$\rightarrow \boxed{D = 2.49 \text{ fm}}$$

$$\phi = \arctan \left( \frac{17}{2.49 \times 10^3} \right)$$

$$= \boxed{+179.6^\circ}$$

